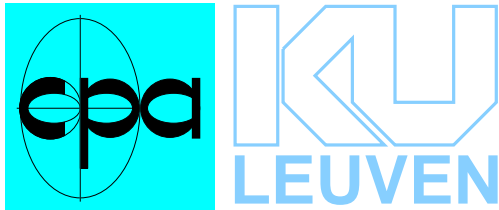

Overcompressive shocks in two-dimensional and three-dimensional magnetohydrodynamic bow shock flows

Hans De Sterck



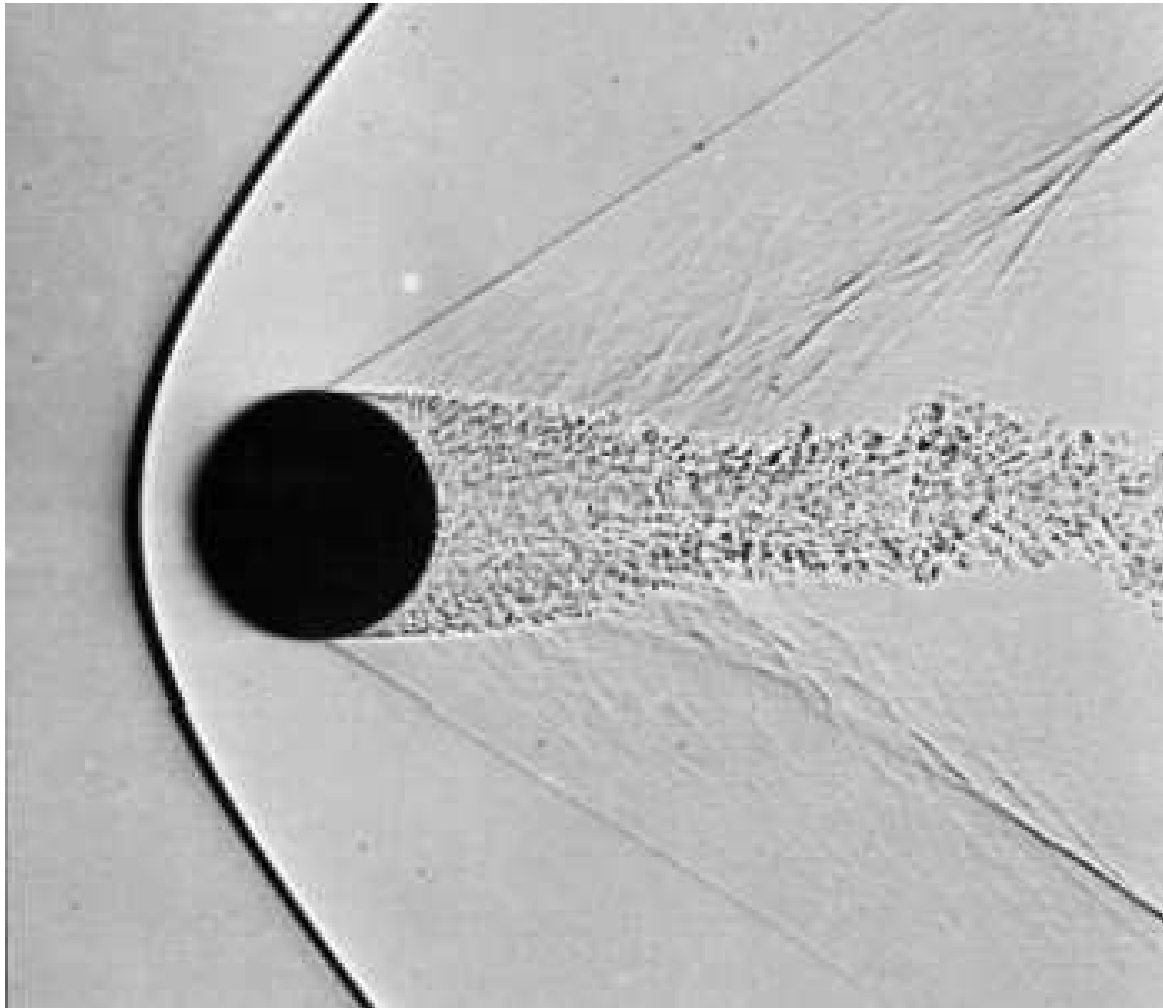
von Karman Institute for Fluid Dynamics, Belgium



Centre for Plasma Astrophysics, K.U. Leuven, Belgium

Stefaan Poedts, Centre for Plasma Astrophysics, K.U. Leuven, Belgium

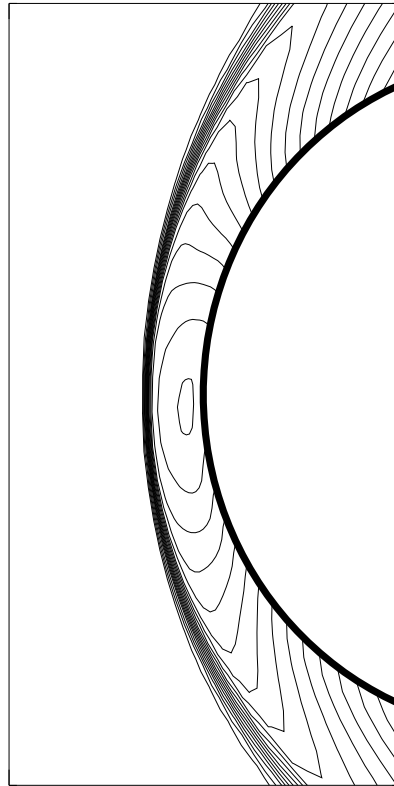
B.C. Low, High Altitude Observatory, NCAR, Boulder, CO, USA



- supersonic flow of air over sphere ($M=1.53$)
- regular bow shock
- (An album of fluid motion, Van Dyke)

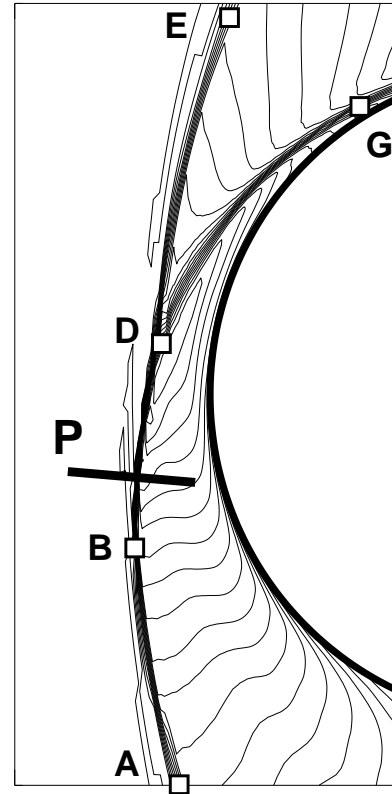
3D MHD bow shock fbws over conducting sphere

(a) pressure-dominated

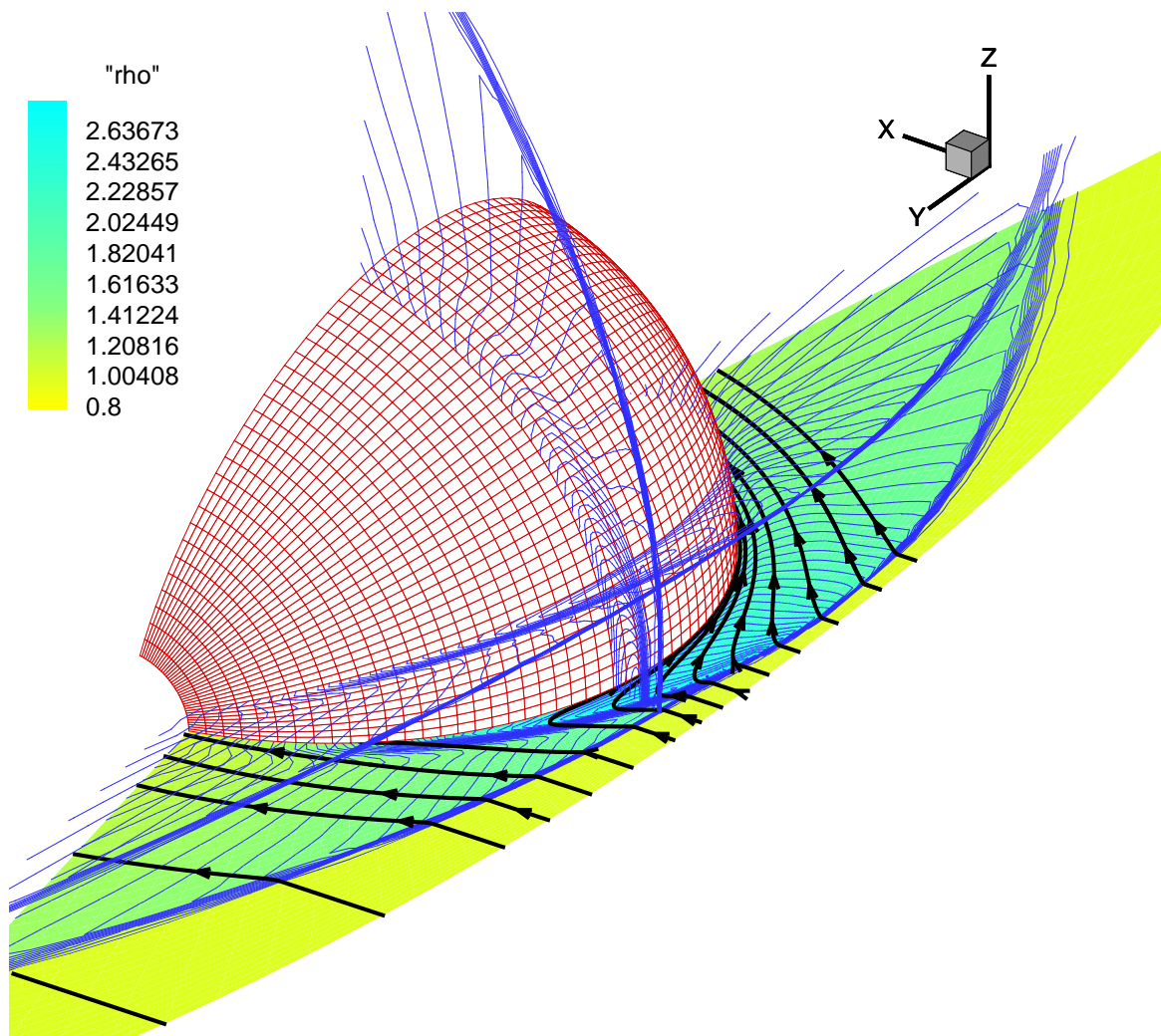


B weak
pressure-dominated

(b) magnetically dominated

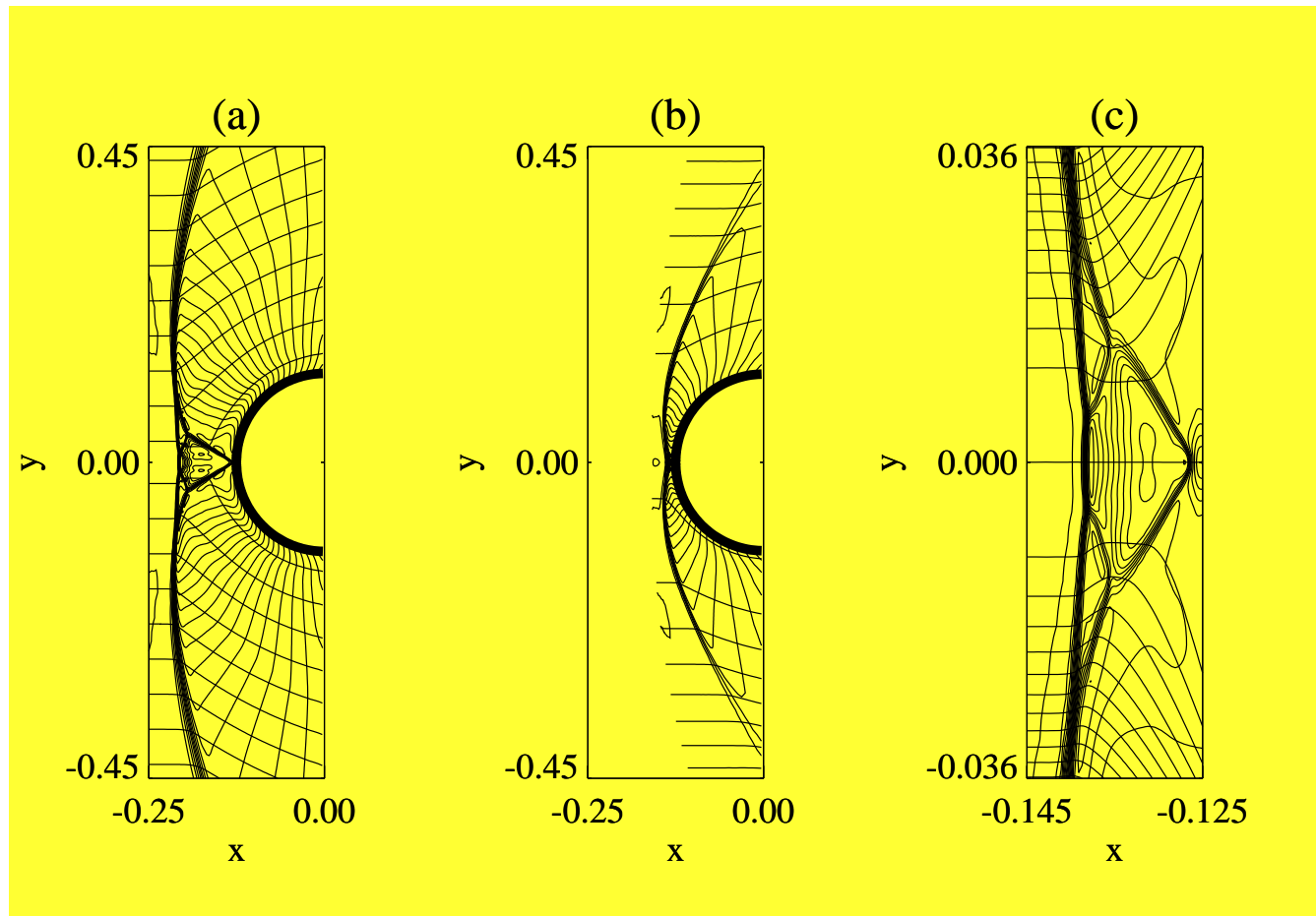


B strong
magnetically dominated



- strong upstream B
- dual-front bow shock
- overcompressive shock segments

Symmetric 2D MHD bow shock fbws over cylinder



2D flow over cylinder, and axisymmetric flow over sphere

- overcompressive shocks, compound shocks
-

Conservative form ideal MHD equations

$$\frac{\partial}{\partial t} \begin{bmatrix} \rho \\ \rho \vec{v} \\ \frac{\rho v^2}{2} + \rho e + \frac{B^2}{2} \\ \vec{B} \end{bmatrix} + \nabla \cdot \begin{bmatrix} \rho \vec{v} \\ \rho \vec{v} \vec{v} + \left(p + \frac{B^2}{2} \right) \vec{I} - \vec{B} \vec{B} \\ \left(\frac{\rho v^2}{2} + \rho e + p \right) \vec{v} - (\vec{v} \times \vec{B}) \times \vec{B} \\ \vec{v} \vec{B} - \vec{B} \vec{v} \end{bmatrix} = 0$$

Mathematical nature MHD equations

- system of **nonlinear hyperbolic** conservation laws:

$$\frac{\partial U}{\partial t} + \frac{\partial F(U)}{\partial x} = 0 \quad \frac{\partial U}{\partial t} + \mathbf{A}(U) \cdot \frac{\partial U}{\partial x} = 0$$

the system is a hyperbolic system of equations



$\mathbf{A}(U)$ has n real eigenvalues and a complete set of eigenvectors $\forall U$



the system has n real characteristic curves

$$\Rightarrow \mathbf{A}(U) = \mathbf{R}(U) \cdot \mathbf{\Lambda}(U) \cdot \mathbf{L}(U)$$

● nonlinear hyperbolic system: \Rightarrow waves, shocks

● HD (Euler): ($n = 5$)

- $\lambda = u, u, u, u + c, u - c$

- one nonlinear wave mode

- isotropic

- one type of shock

● MHD: ($n = 8$)

- $\lambda = u, u, u + c_f, u - c_f,$

$u + c_A, u - c_A, u + c_s, u - c_s$

- three wave modes: fast, Alfvén, slow

- strongly anisotropic

- three types of shocks

● hyperbolic theory of MHD:

- non-strictly hyperbolic

- non-convex \Rightarrow compound shocks

- rotationally invariant \Rightarrow instability of (overcompressive) intermediate shocks

Overview

1) MHD waves and shocks

2) Topology of bow shock flows

3) Non-convexity: compound shocks

4) Overcompressive shocks

5) 'Every constant state is bordered by a simple wave'

1) MHD waves and shocks

1) MHD waves and shocks

MHD waves

- three anisotropic waves

$$c_{fx} \geq c_{Ax} \geq c_{sx}$$

- non-strictly hyperbolic

- x along \vec{B} : $c_{fx} = c_{Ax}$ or $c_{Ax} = c_{sx}$

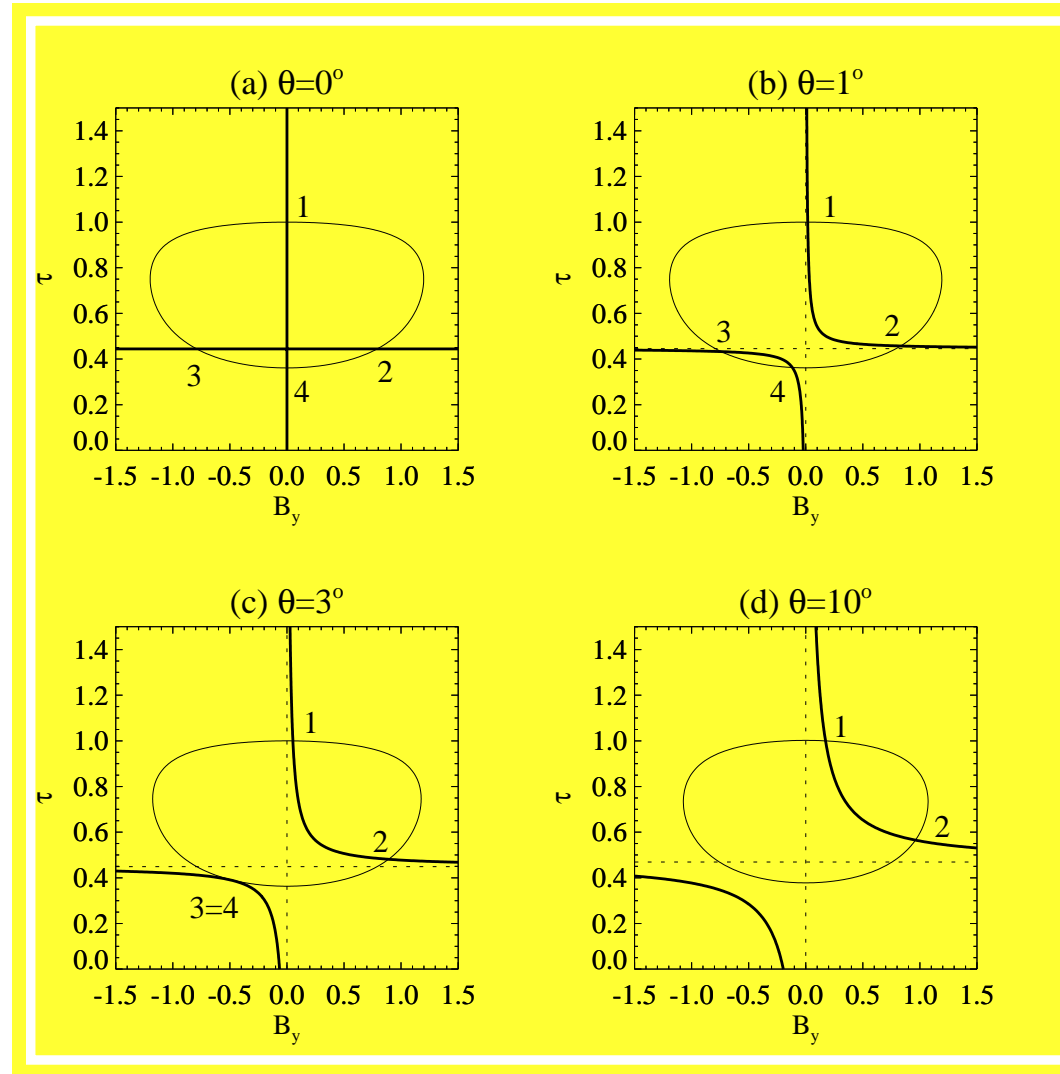
(exceptionally $c_{fx} = c_{Ax} = c_{sx}$)

- x perpendicular to \vec{B} : $c_{Ax} = c_{sx} = 0$

1) MHD waves and shocks

MHD shock types

- MHD Rankine-Hugoniot relations
- up to 4 fixed points in phase space diagram

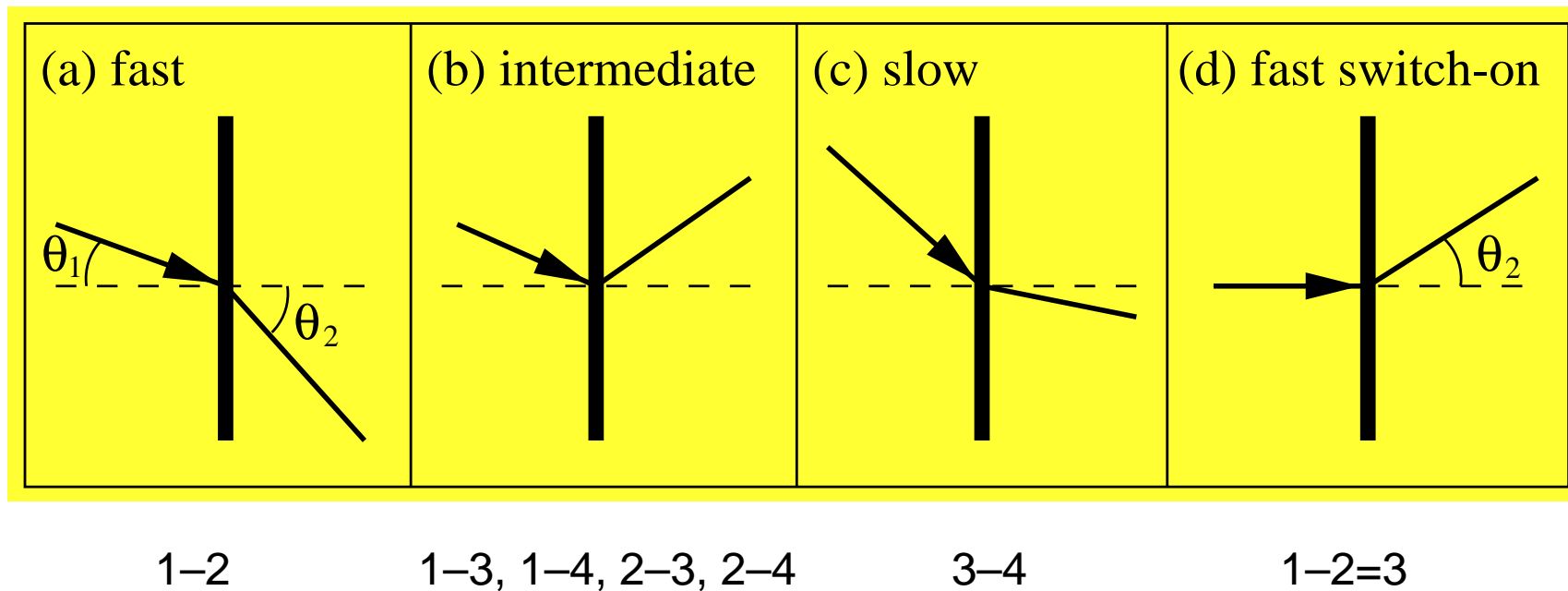


1) MHD waves and shocks

- 4 possible positions of flow speed in direction x :

$$\boxed{1} \geq c_{fx} \geq \boxed{2} \geq c_{Ax} \geq \boxed{3} \geq c_{sx} \geq \boxed{4}$$

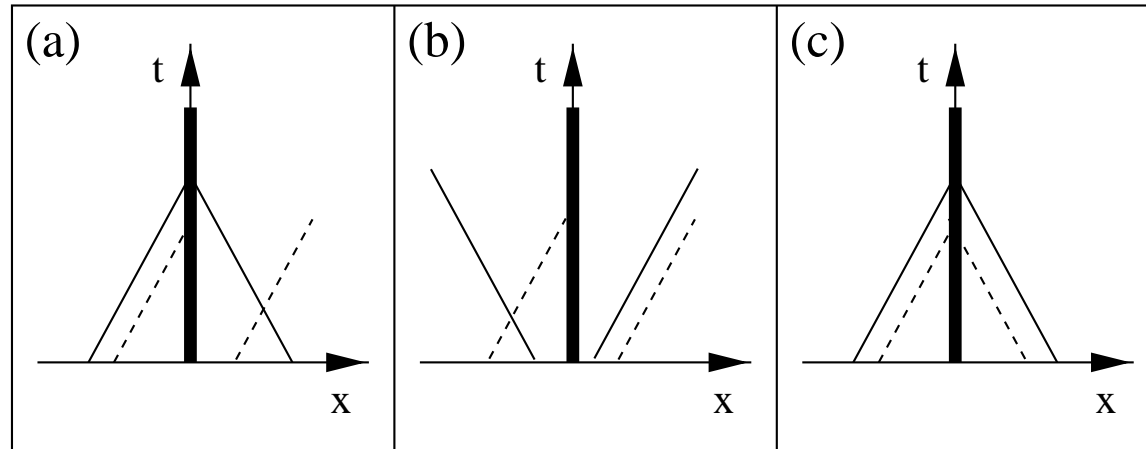
⇒ three types of MHD shocks:



- 1-3, 2-4 and 1-4 intermediate shocks are overcompressive

⇒ instability can arise, see later

1) MHD waves and shocks



(slope of i th characteristic in xt plane = $\lambda_i = v_n - c_{fn}$ or $v_n - c_{An}$ or ...)

- (a) Lax shock: $2n - 1$ characteristics impinging
 - (b) undercompressive shock
 - (c) overcompressive shock
-

1) MHD waves and shocks

- three wave speeds depend on direction

⇒ three Mach numbers too:

- $M_{fx} = \frac{|v_x|}{c_{fx}}$

- $M_{Ax} = \frac{|v_x|}{c_{Ax}}$

- $M_{sx} = \frac{|v_x|}{c_{sx}}$

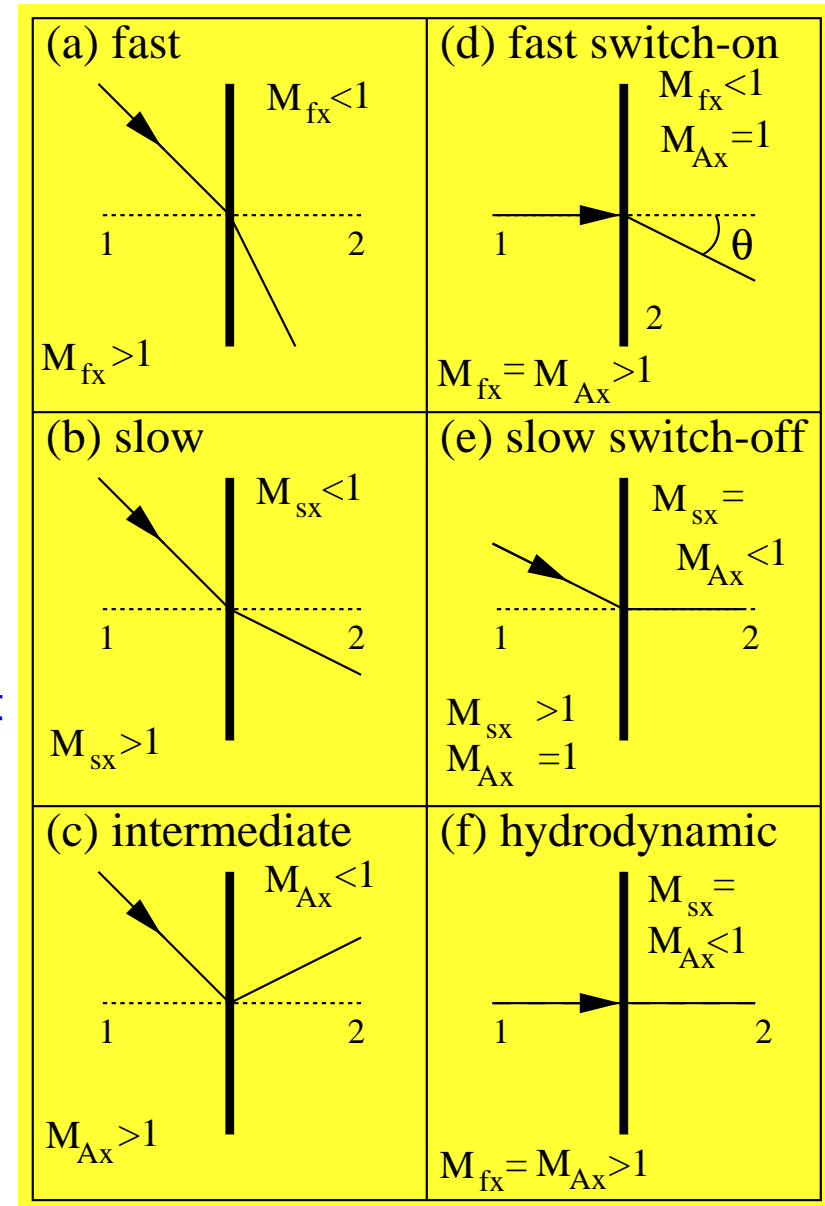
- fast switch-on shocks = **intrinsic magnetic effect**

- only occur when:

- 1) $B_1^2 > \gamma p_1$

- 2) $B_1^2 > \rho_1 v_{x,1}^2 \frac{\gamma - 1}{\gamma(1 - \beta_1) + 1}$

⇒ upstream flow is **magnetically dominated**



Numerical simulation technique

- MHD is hyperbolic, like Euler \Rightarrow use CFD techniques
- finite volume, structured grid
- second order in space (limited slope reconstruction)
- second order in time (explicit two-stage Runge-Kutta)
- parallel on 32 processors using message passing (MPI) (up to 1 000 000 cells)
- $\nabla \cdot \vec{B}$ constraint:

MHD has singular Jacobians, which leads to numerical instabilities

\Rightarrow add source term

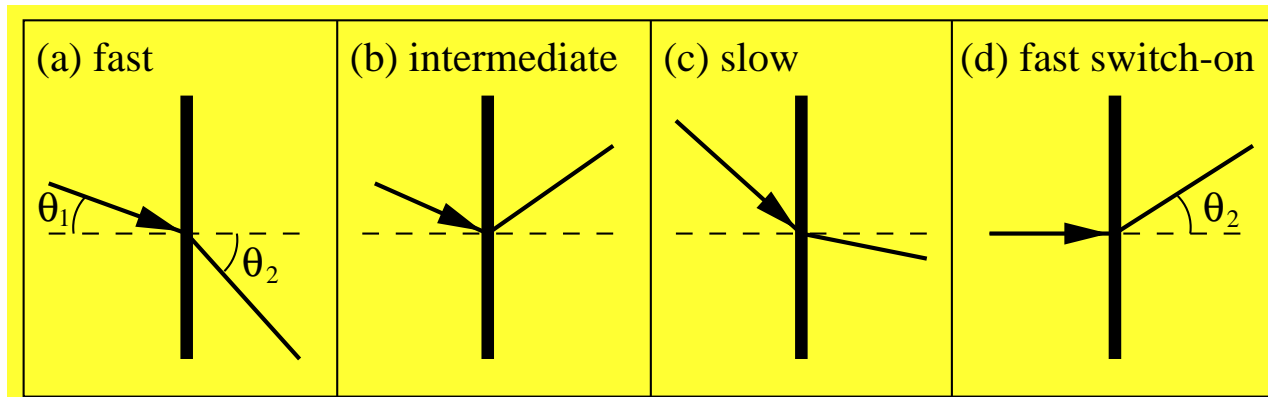
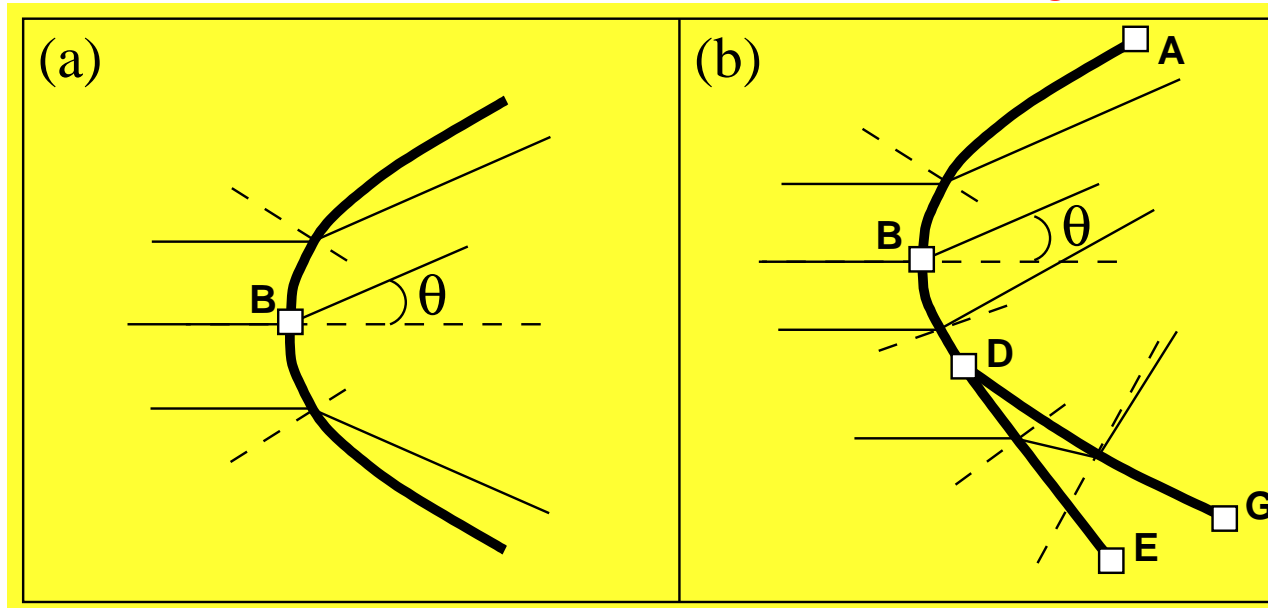
- makes equations symmetrizable
 - makes equations Galilean invariant
 - makes numerical scheme stable
-

2) Topology of bow shock flows

2) Topology of bow shock flows

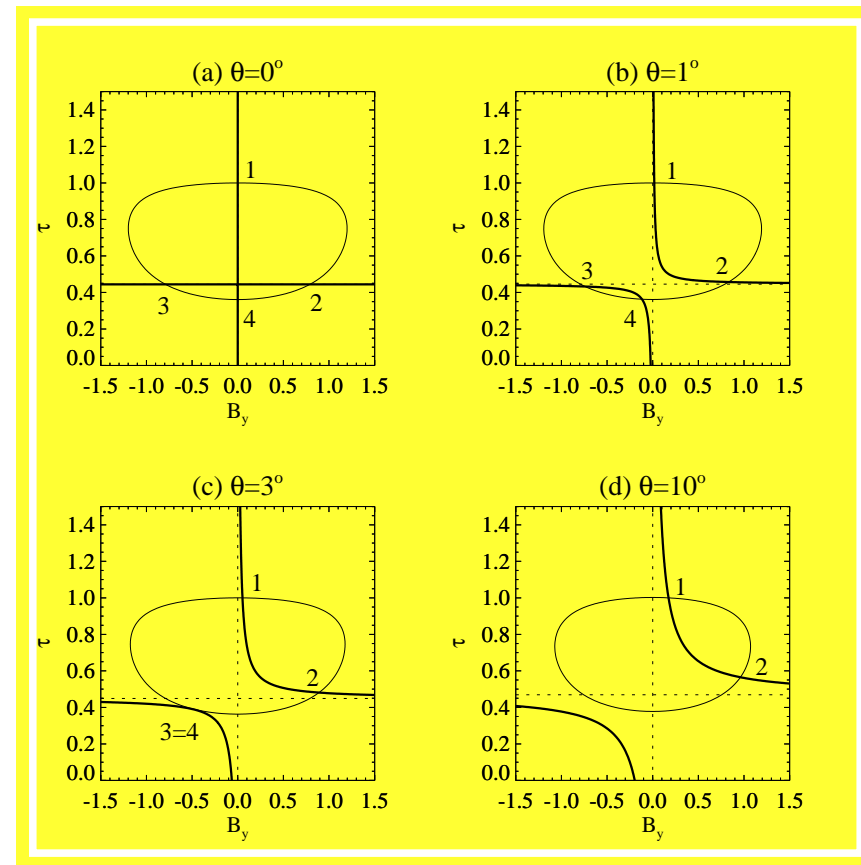
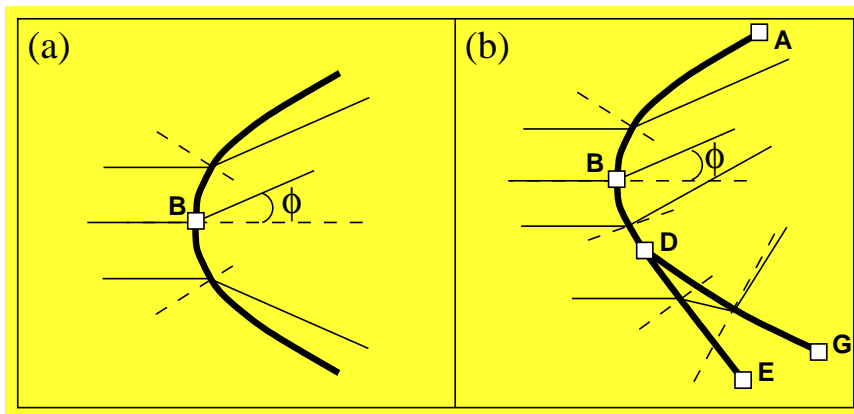
B weak

B strong



2) Topology of bow shock flows

- why does the shock split up at point D?

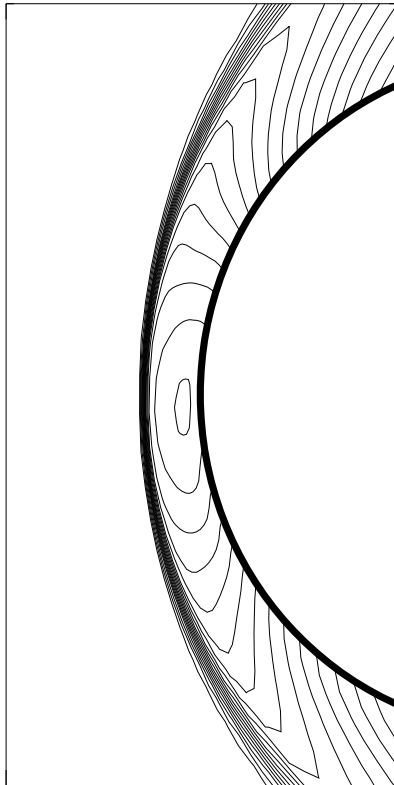


- A–B, D–E: fast 1–2
- B–D: intermediate 1–3
- D–G: intermediate 2–4 evolving into slow 3–4

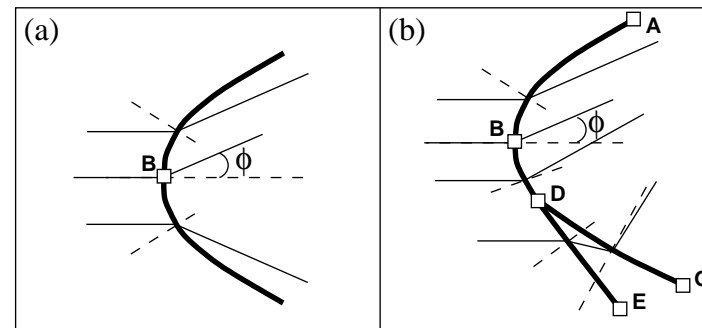
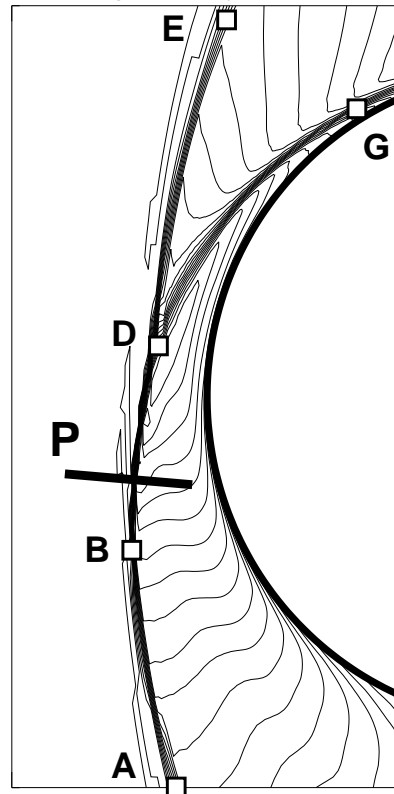
2) Topology of bow shock flows

this topology in 3D MHD bow shock flows

(a) pressure-dominated

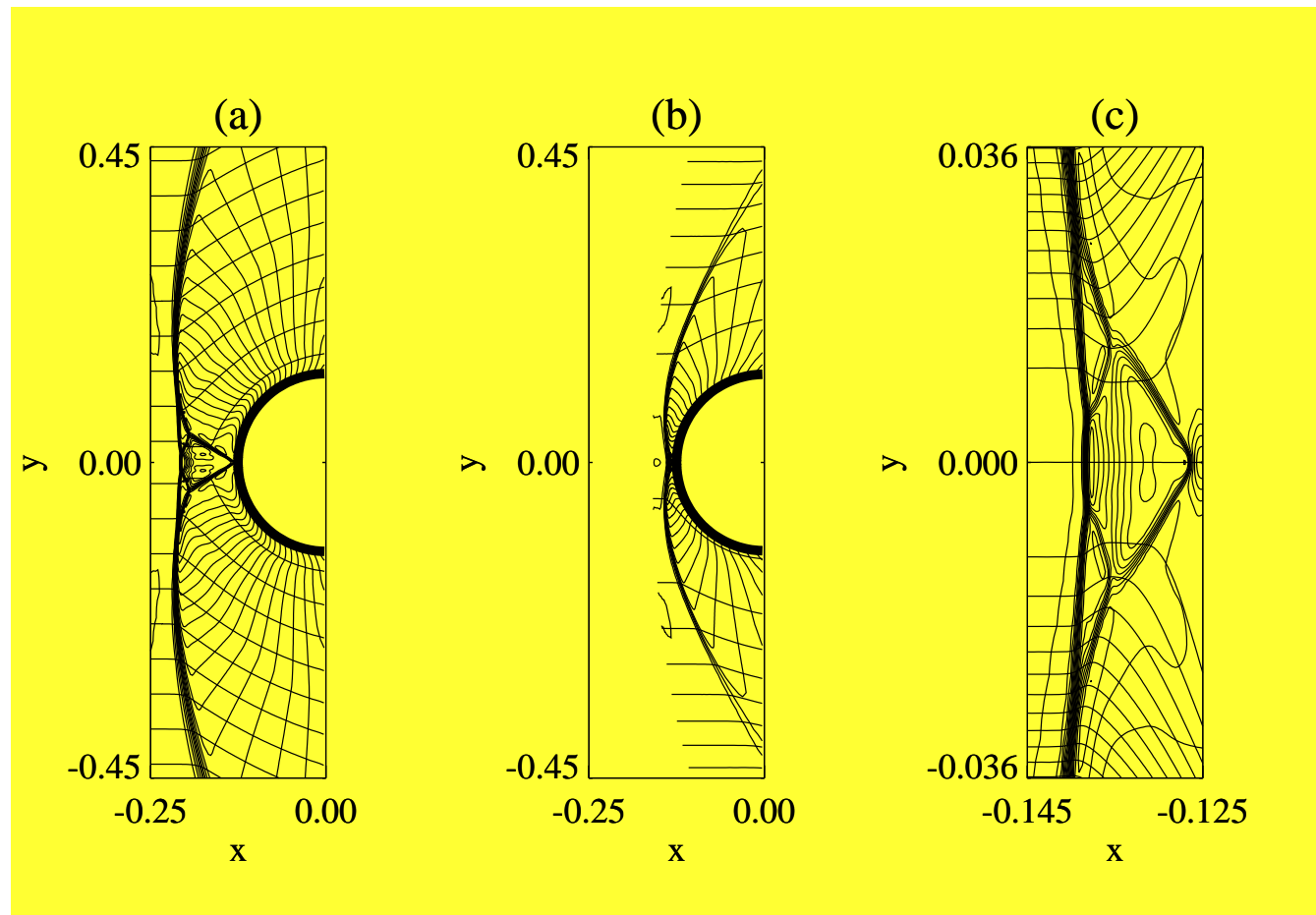


(b) magnetically dominated



2) Topology of bow shock flows

this topology also in 2D MHD bow shock flows



2D flow over cylinder, and axisymmetric flow over sphere

3) Non-convexity: compound shocks

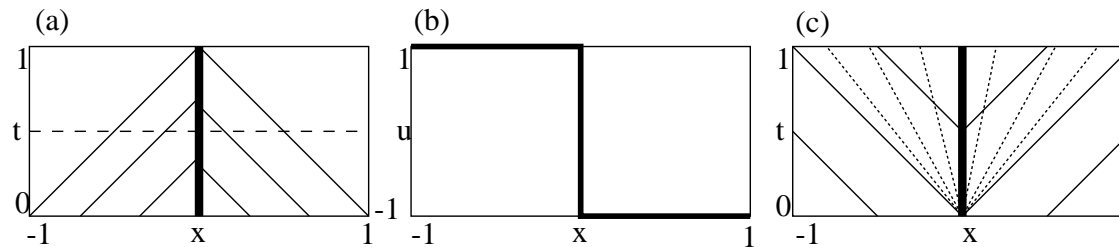
3) Non-convexity: compound shocks

$$\left(\frac{\partial u}{\partial t} + f'(u) \frac{\partial u}{\partial x} = 0 \right)$$

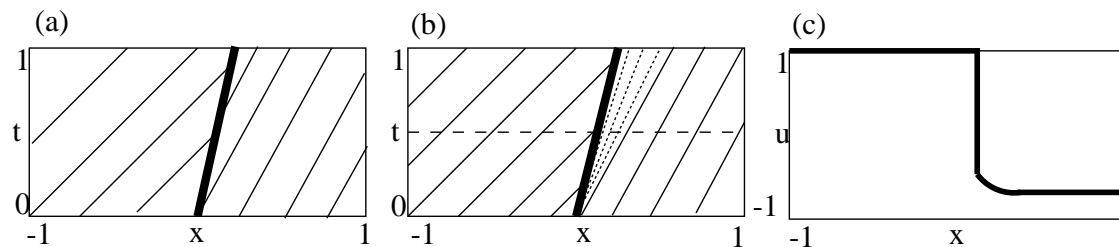
f is convex $\Leftrightarrow f'$ is monotone $\Leftrightarrow f''$ does not change sign

• f is convex ($f = u^2$, Burgers, Euler) \Rightarrow

$$f'(u_1) \geq s = (f(u_2) - f(u_1))/(u_2 - u_1) \geq f'(u_2)$$



• f is non-convex ($f = u^3$) $\Rightarrow f'(u_1) \geq f'(u_2) \geq s$



compound shock: on rarefaction side: sonic ($M = 1$), tangent characteristic

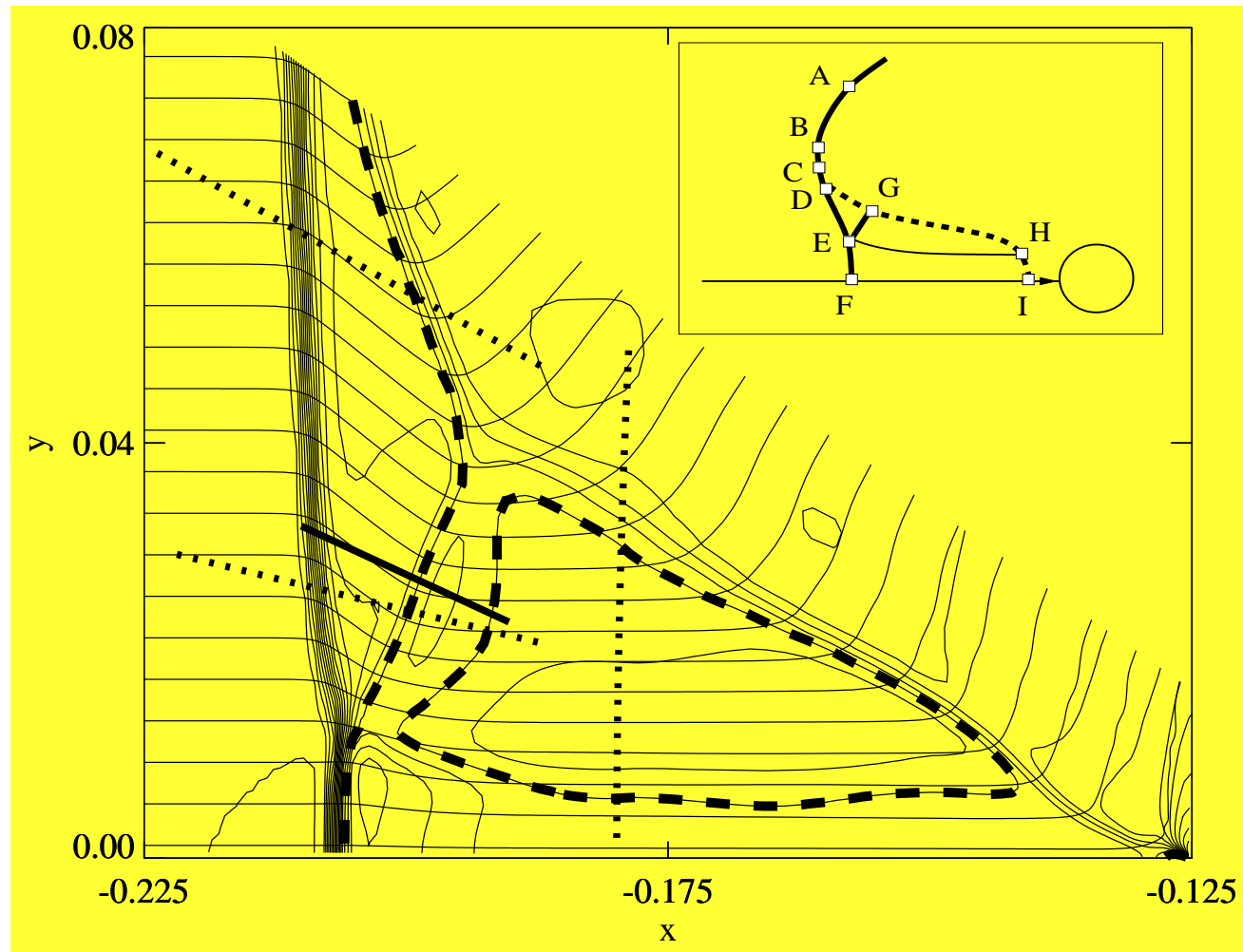
3) Non-convexity: compound shocks

$$\boxed{1} \geq c_{fx} \geq \boxed{2} \geq c_{Ax} \geq \boxed{3} \geq c_{sx} \geq \boxed{4}$$

- c_f, c_s non-convex :
 - 1=2-3 and 2-3=4 sonic shocks in compound shocks
 - 1=2-3=4 double compound shock

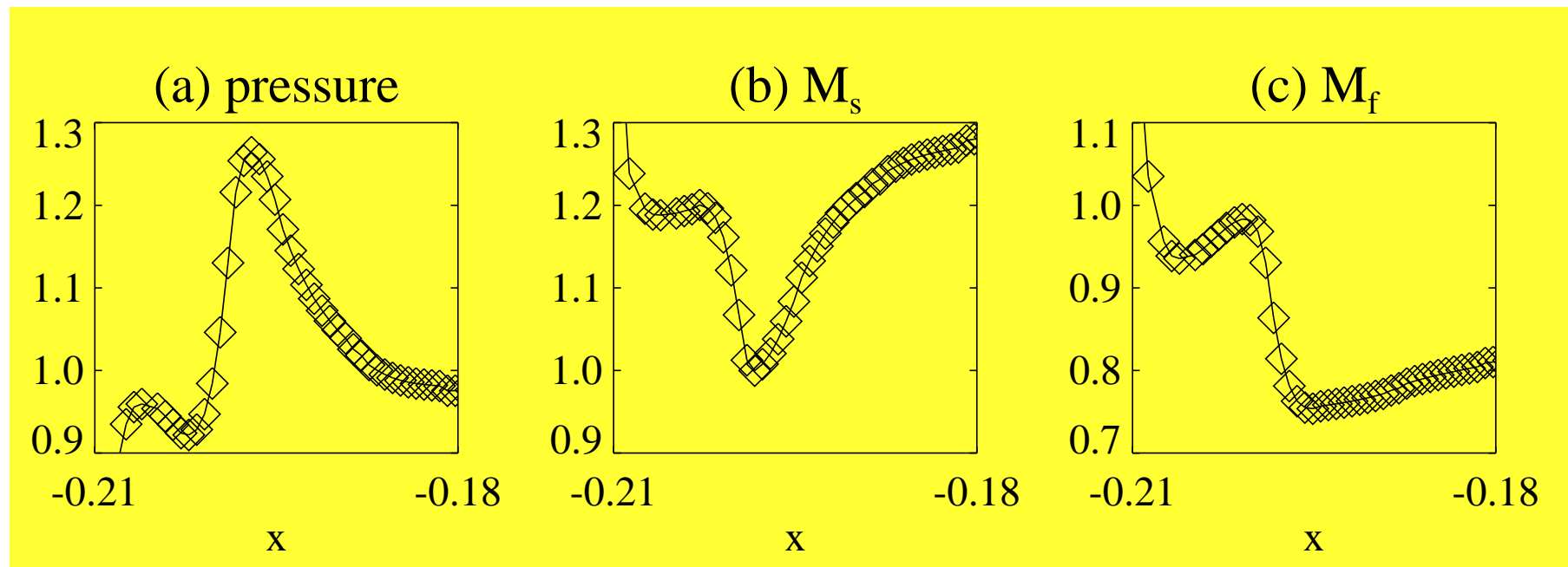
3) Non-convexity: compound shocks

compound shocks in 2D fbw



Magnetic field lines and M_A contour lines

3) Non-convexity: compound shocks



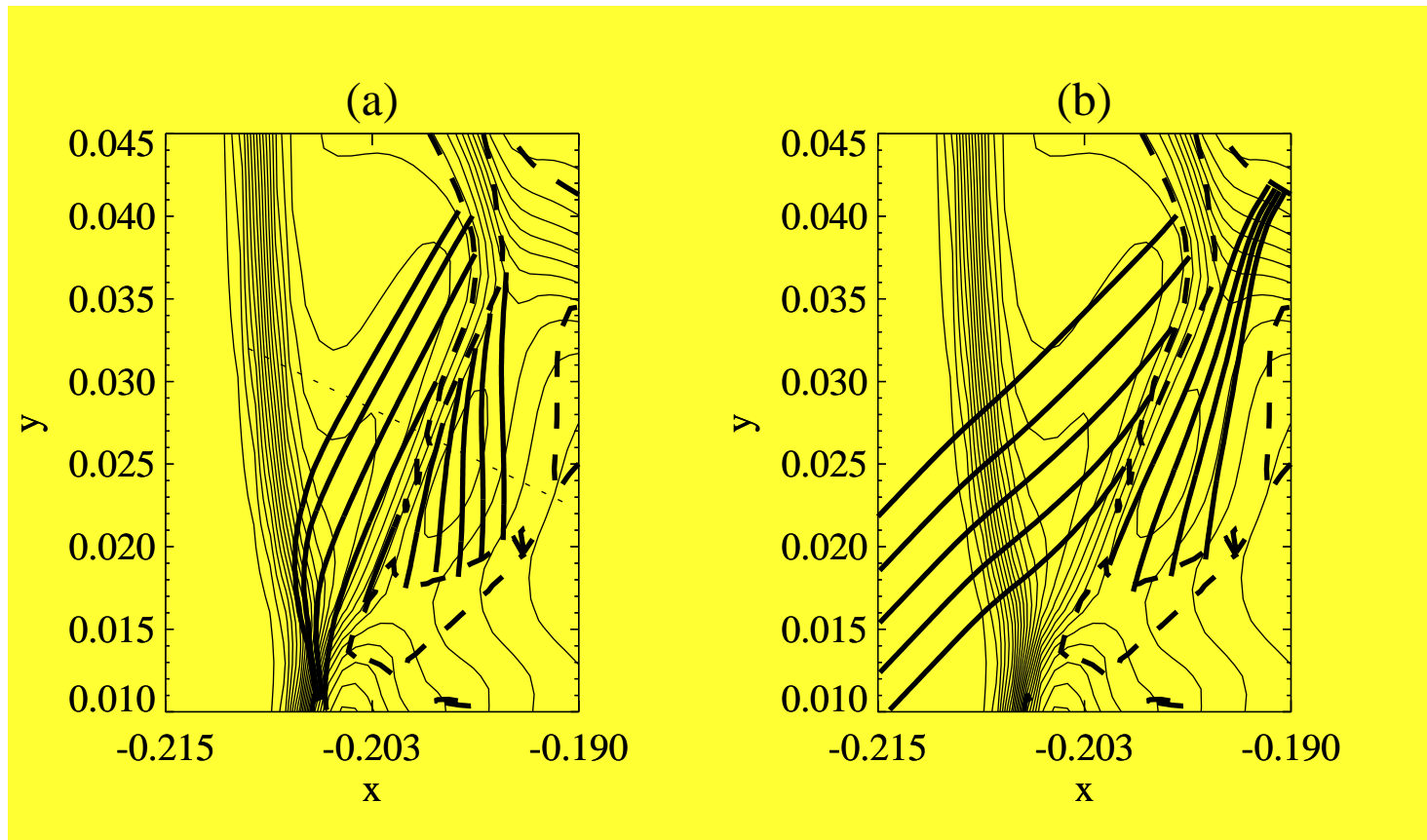
Cut along solid line

- E-G shock is preceded and followed by rarefaction regions
- $M_f = 1$ where upstream (left) rarefaction is attached to shock
- $M_s = 1$ where downstream (right) rarefaction is attached to shock

\Rightarrow E-G: 1=2-3=4 shock

\Rightarrow stationary double compound shock!

3) Non-convexity: compound shocks



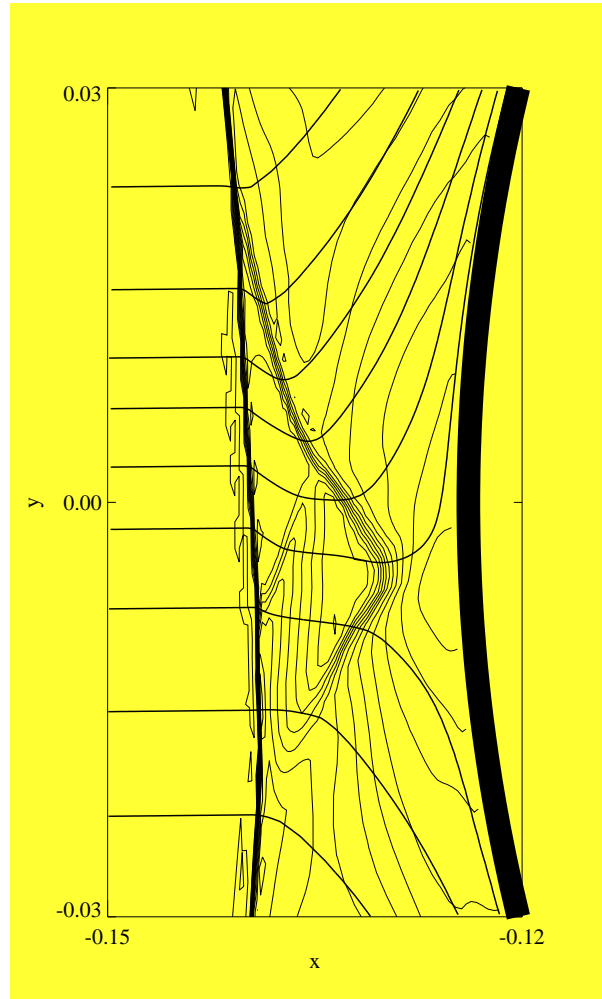
Two families of characteristics near E-G shock

⇒ characteristic analysis ⇒ **stationary double compound shock!**

⇒ xy mathematical equivalent of known xt double compound shocks

3) *Non-convexity: compound shocks*

compound shocks in 3D fbw



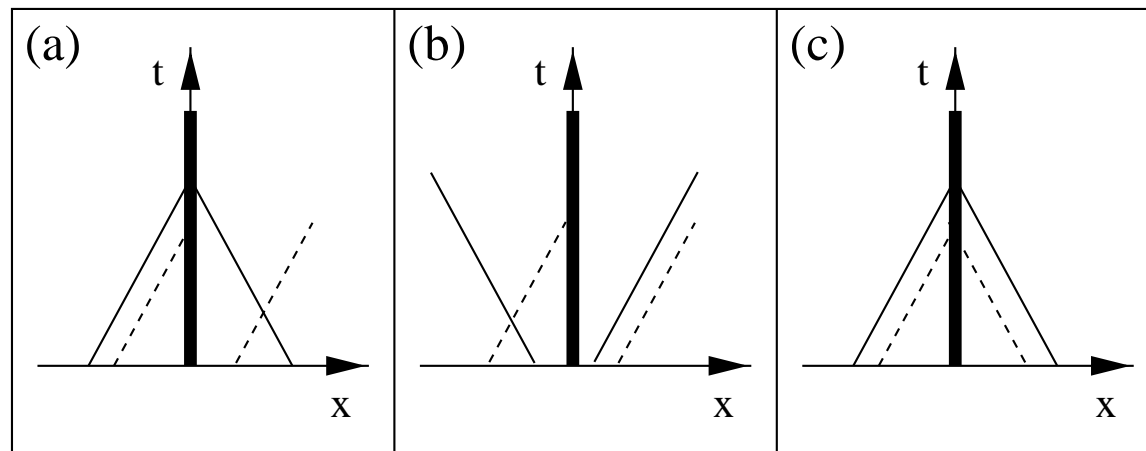
- conclusion: effects of non-convexity of MHD in 2D and 3D flows
-

4) Overcompressive shocks

4) Overcompressive shocks

2×2 system ($n = 2$)

$$\frac{\partial}{\partial t} \begin{bmatrix} u \\ v \end{bmatrix} + \frac{\partial}{\partial x} \begin{bmatrix} f(u, v) \\ g(u, v) \end{bmatrix} = 0$$



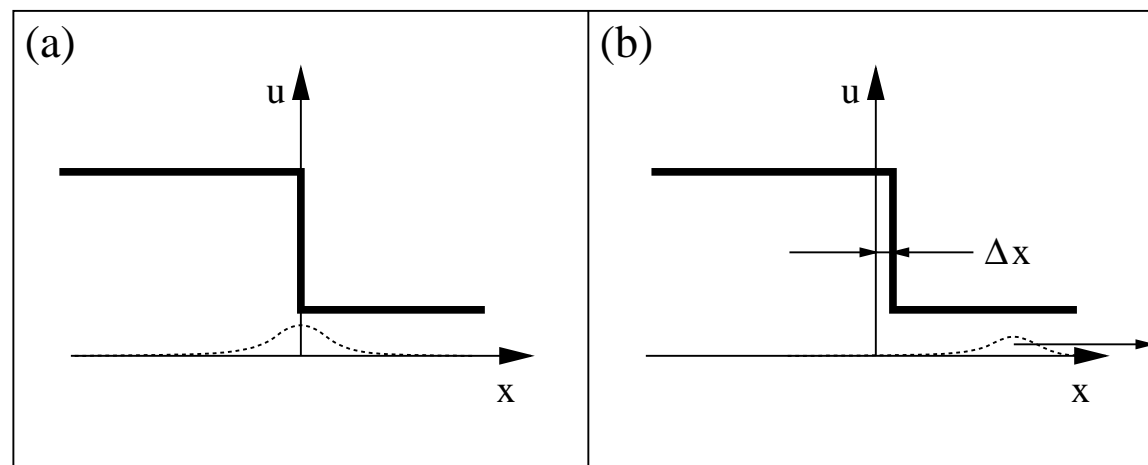
- (a) Lax shock: $2n - 1$ characteristics impinging
 - (b) undercompressive shock
 - (c) overcompressive shock
-

4) Overcompressive shocks

Strictly hyperbolic system

$$\frac{\partial}{\partial t} \begin{bmatrix} u \\ v \end{bmatrix} + \frac{\partial}{\partial x} \begin{bmatrix} v \\ v^2/u + c^2 u \end{bmatrix} = 0$$

- isothermal Euler
- $\lambda = \frac{v}{u} \pm c$
- perturbation: $\int u dx$ and $\int v dx$ remain constant
- shock is displaced, wave is sent out



4) *Overcompressive shocks*

- ideal (hyperbolic) system:

- Lax shock is stable against perturbations

- dissipative (parabolic) system:

- unique viscous profile connects unstable node with saddle
- Lax shock has stable dissipative (viscous) profile

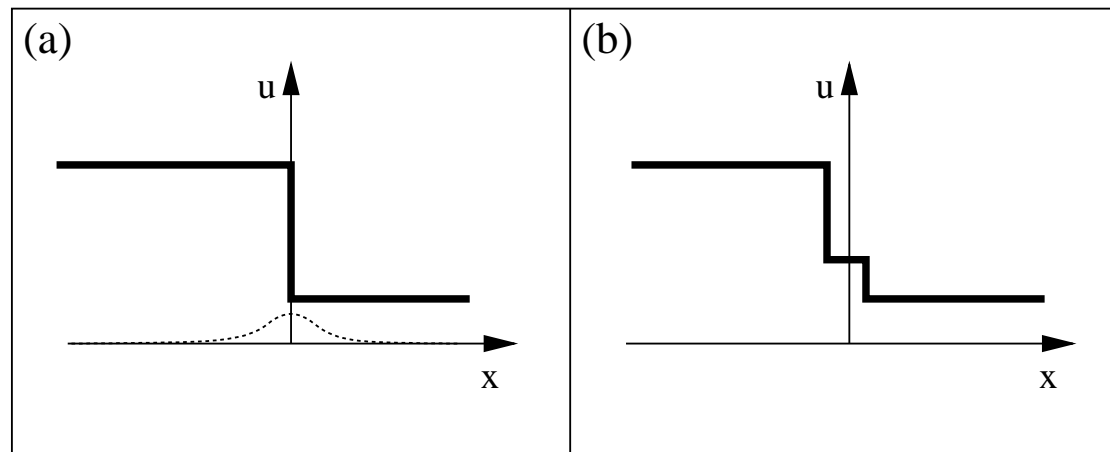
⇒ vanishing viscosity limit:

- shock admissibility is the same in ideal and dissipative system
-

4) Overcompressive shocks

System with overcompressive shocks: Type A, non-rotational

$$\frac{\partial}{\partial t} \begin{bmatrix} u \\ v \end{bmatrix} + \frac{\partial}{\partial x} \begin{bmatrix} a u^2/2 + b v \\ v^2/2 \end{bmatrix} = 0$$



- perturbation: $\int u dx$ and $\int v dx$ remain constant
- shock can be displaced, wave CANNOT be sent out: problem!

4) *Overcompressive shocks*

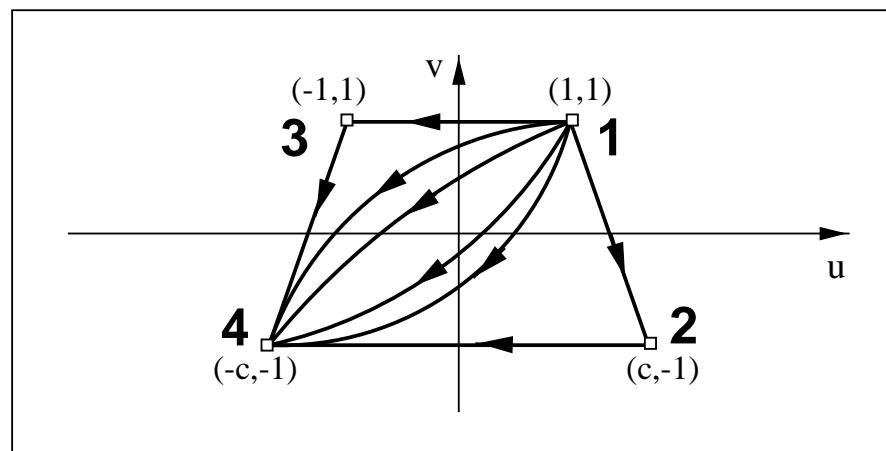
- ideal (hyperbolic) system:

- overcompressive shock splits under generic perturbation = unstable
- BUT: resulting shocks have same speed as original shock = L1 stable

- dissipative (parabolic) system:

- family of viscous profiles connecting unstable node with stable node
- upon perturbation, viscous shock is translated, and different profile is assumed
- viscous shock is stable under arbitrary perturbation

⇒ ideal \neq dissipative !! (vanishing viscosity limit?)

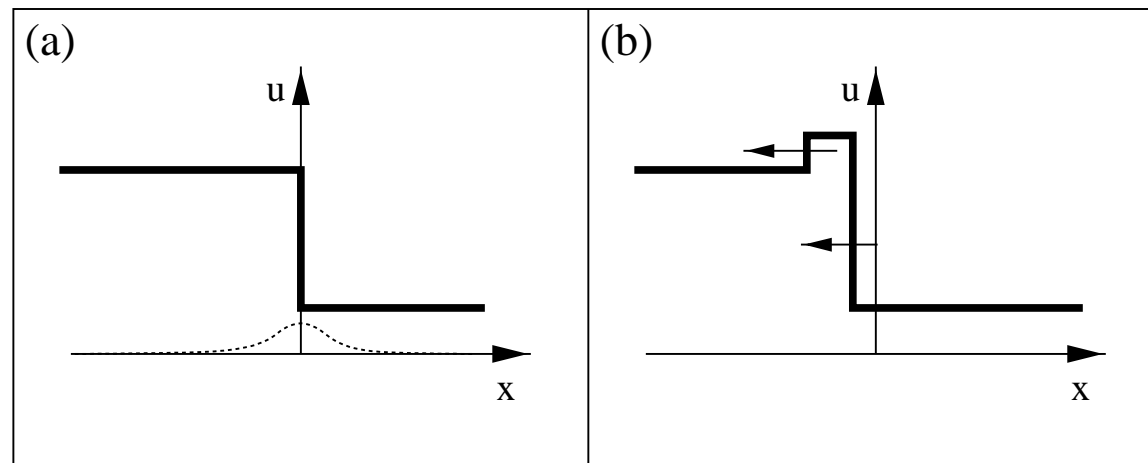


4) Overcompressive shocks

System with overcompressive shocks: Type B, rotational

$$\frac{\partial}{\partial t} \begin{bmatrix} u \\ v \end{bmatrix} + \frac{\partial}{\partial x} \begin{bmatrix} u(u^2 + v^2) \\ v(u^2 + v^2) \end{bmatrix} = 0$$

- rotational invariance: $F(U) = \Phi(\|U\|^2)U$
- perturbation: $\int u dx$ and $\int v dx$ remain constant
- shock can be displaced, wave CANNOT be sent out: problem!



4) Overcompressive shocks

- ideal (hyperbolic) system:

- overcompressive shock splits under generic perturbation = unstable
- AND: resulting shocks have speed different from original shock = not L1 stable
- generically unstable

- dissipative (parabolic) system:

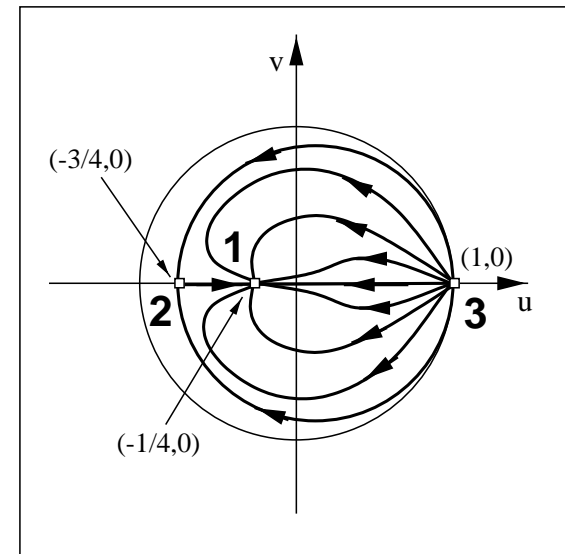
- family of viscous profiles connecting unstable node with stable node

- upon perturbation, viscous shock is translated, and different profile is assumed

- BUT: if $\int v dx > M_{crit}$, shock disintegrates
- BUT: M_{crit} vanishes for vanishing dissipation

⇒ viscous shock is conditionally stable

⇒ ideal \neq dissipative !! (vanishing viscosity limit???)



4) Overcompressive shocks

Overcompressive shocks in MHD

$$\boxed{1} \geq c_{fx} \geq \boxed{2} \geq c_{Ax} \geq \boxed{3} \geq c_{sx} \geq \boxed{4}$$

- **planar MHD** (2D)

$B_z \equiv v_z \equiv 0$, c_A drops out

- 1–3 and 2–4 NOT overcompressive, but **Lax**

- 1–4 **overcompressive** (Type A)

- **full MHD**

- 1–3 and 2–4 **overcompressive** (Type B)

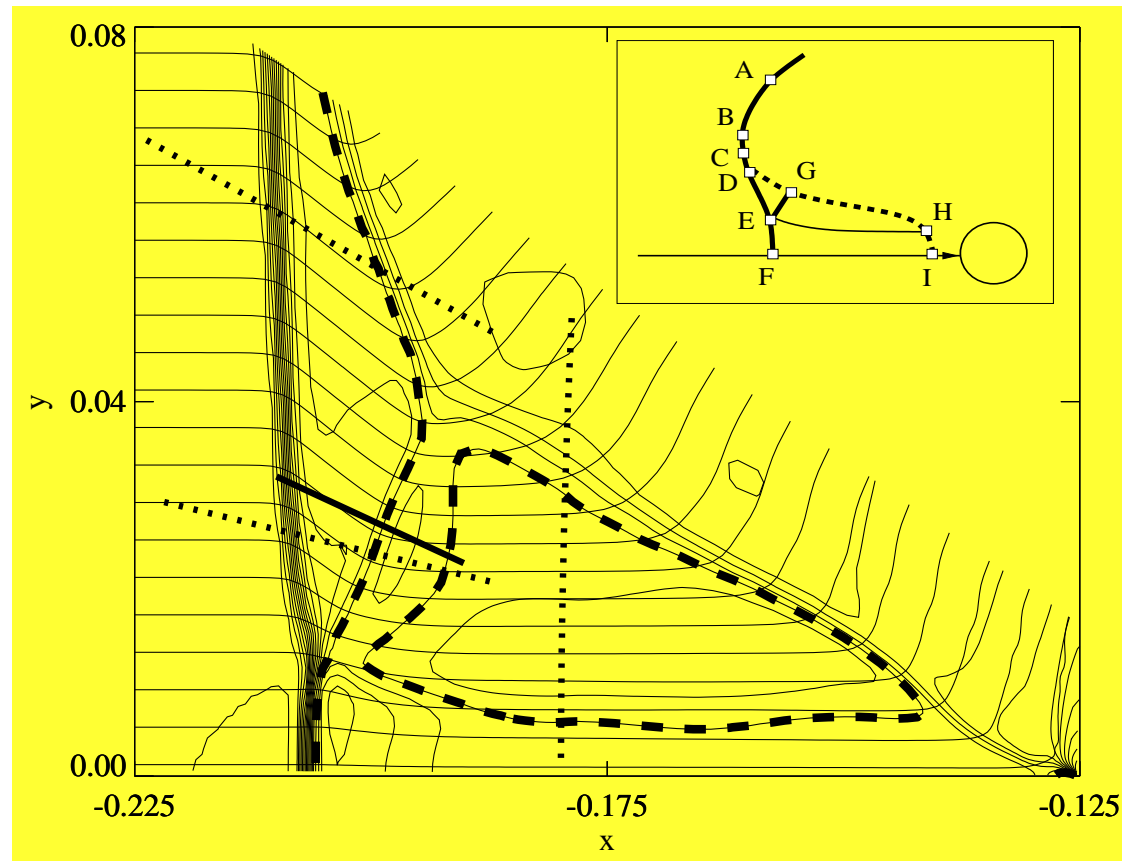
- 1–4 **overcompressive** (Type B)

- complication: bifurcation in existence of viscous profiles as function of the dissipative parameters

⇒ intermediate shocks (1–3, 2–4 and 1–4) can arise in MHD flows for small dissipation

4) Overcompressive shocks

2D bow shock: intermediate shocks



D-E: fast shock (almost switch-on)

E-F: 1-4 intermediate shock

at F: 1-4 hydrodynamic shock

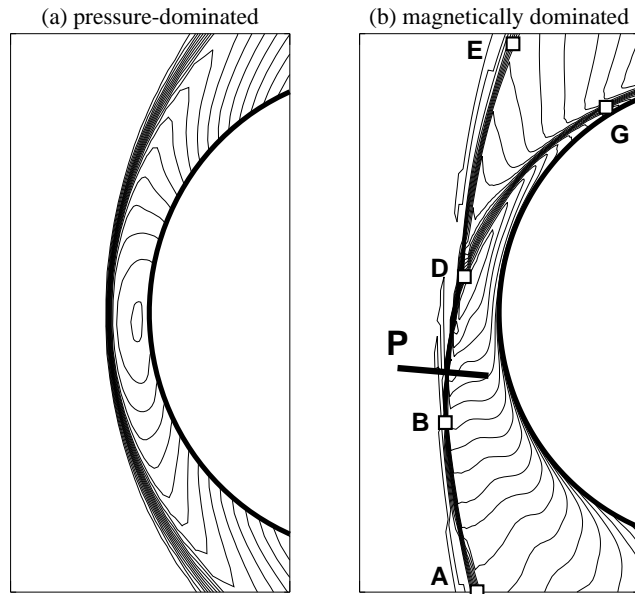
E-H: tangential discontinuity

E-G: intermediate shock

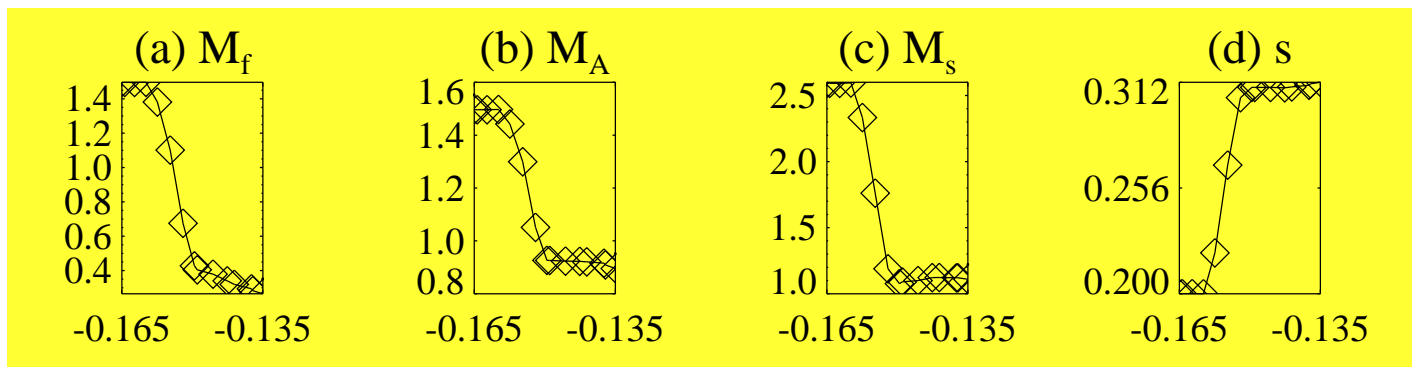
D-G-H-I: 2-4 and 2=3-4 intermediate shock

4) Overcompressive shocks

3D bow shock: intermediate 1–3 shock in the leading front



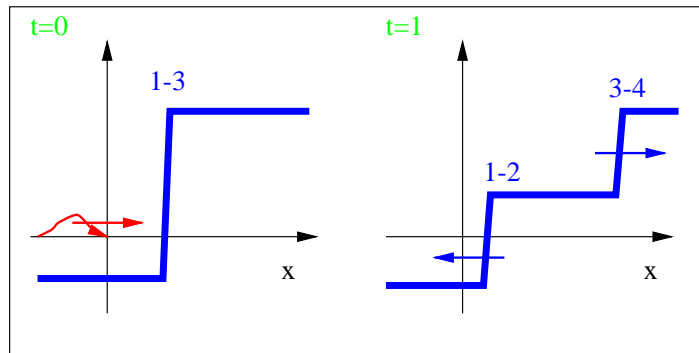
- along cut P: fast and Alfvénic Mach number pass 1



4) Overcompressive shocks

Large-scale stability of 3D bow shock flows

with intermediate shock segments



● **question:** can an intermediate shock survive in a plasma where perturbations are present?

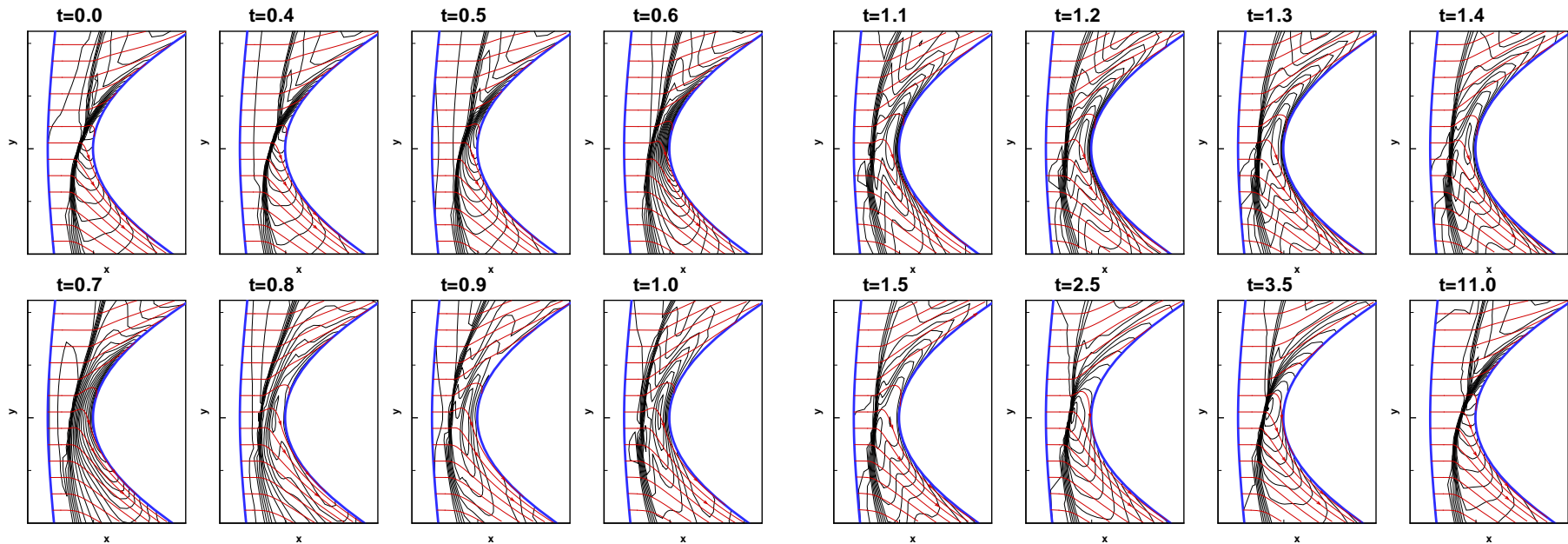
first thought: no, the perturbations will destabilize the shock

second thought: YES: in DRIVEN problems an intermediate shock can be *reformed* (through nonlinear steepening) after the perturbation has passed

● this is proved in the following numerical experiments which are perturbations of the above described 3D stationary flows

4) Overcompressive shocks

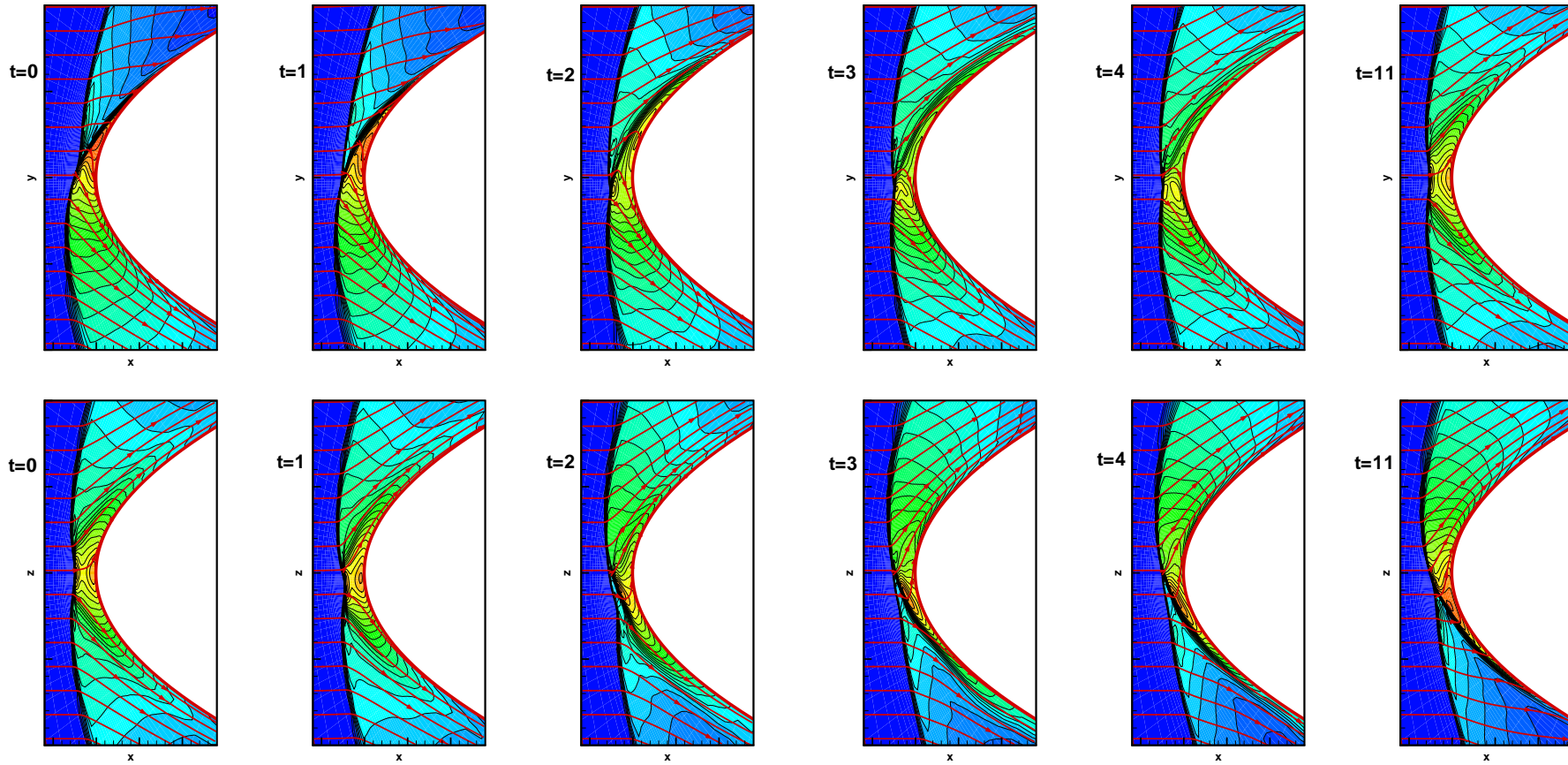
- **EXPERIMENT A:** perturb $B_z = \exp(-(t - 0.5)^2/0.04)$



\Rightarrow the 1–3 intermediate shock segment splits up in other shocks, but is **REFORMED** when the perturbation has passed

4) Overcompressive shocks

- **EXPERIMENT B:** rotate v around the x -axis from $v_z = 0$ to $v_y = 0$ between $t = 0$ and $t = 1$



⇒ the 1–3 intermediate shock segment in the xy plane splits up into two other shocks, but is **REFORMED** at a different location – in the xz plane – corresponding to the new (steady) inflow conditions

4) *Overcompressive shocks*

remaining questions

- what is the ideal MHD solution for the 3D bow shock problem?
 - (without intermediate (overcompressive) shocks?)
 - what is the solution for ranges of dissipative parameters for which some intermediate shocks do not exist?
 - in physical plasmas: can intermediate shocks be observed?
 - will the supercritical perturbations be infrequent enough such that the intermediate shocks get time to form ? (turbulence)
 - what are the dissipation mechanisms and coefficient magnitudes?
 - simulations with explicit discretization of dissipative terms
 - (huge parameter space)
-

5) 'Every constant state is bordered by a simple wave'

5) 'Every constant state is bordered by a simple wave'

Simple waves

$$\frac{\partial U}{\partial t} + \frac{\partial F(U)}{\partial x} = 0 \quad n \times n \text{ system}$$

Solutions $U(x, t)$ that are constant on curves $x(t)$ in a region of the (x, t) plane

$$\text{simple wave} \Leftrightarrow \frac{dU(x(t), t)}{dt} \equiv 0$$

(U constant on curves $x(t)$)

5) 'Every constant state is bordered by a simple wave'

Simple waves in systems

$$\frac{\partial U}{\partial t} + F'(U) \cdot \frac{\partial U}{\partial x} = 0 \quad \text{and curve } x(t) : \frac{\partial x(t)}{\partial t} = \xi$$

$$\text{simple wave: } \frac{dU(x(t), t)}{dt} = \frac{\partial U(x(t), t)}{\partial t} + \frac{\partial U(x(t), t)}{\partial x} \frac{\partial x(t)}{\partial t} \equiv 0$$

$$\frac{\partial U(x(t), t)}{\partial t} + \frac{\partial U(x(t), t)}{\partial x} \xi = 0$$

$$F'(U) \cdot \frac{\partial U(x(t), t)}{\partial t} + F'(U) \cdot \frac{\partial U(x(t), t)}{\partial x} \xi = 0 \quad \Rightarrow$$

$$F'(U) \cdot \frac{\partial U(x(t), t)}{\partial t} = \xi \frac{\partial U(x(t), t)}{\partial t}$$

5) 'Every constant state is bordered by a simple wave'

$$F'(U) \cdot \frac{\partial U(x(t), t)}{\partial t} = \xi \frac{\partial U(x(t), t)}{\partial t}$$

$$\Rightarrow \xi = \lambda_k(u)$$

simple wave of the k th family

curve $x(t)$ is characteristic curve, and because U is constant on $x(t)$, the k -characteristic is a straight line

$$\Rightarrow \frac{\partial U(x(t), t)}{\partial t} = \alpha R_k$$

time derivative proportional to k th right eigenvector

$$\Rightarrow \frac{\partial U(x(t), t)}{\partial x} = -\frac{1}{\xi} \frac{\partial U(x(t), t)}{\partial t} = \beta R_k$$

space derivative proportional to k th right eigenvector

\Rightarrow leads to equivalent definition of simple wave:

simple wave of k th family $\Leftrightarrow U$ is constant on k -characteristic (straight line)

\Rightarrow only k th wave mode is 'active' in simple wave of k th family

5) 'Every constant state is bordered by a simple wave'

k -Riemann Functions (k -RF)

for a given k , k -Riemann Functions are scalar functions

$$v_s^{(k)}(U) \quad s = 1 \dots n - 1$$

with property

$$\frac{\partial v_s^{(k)}(U)}{\partial U} \cdot R_k(U) = 0$$

$$(\nabla v_s^{(k)}(U)) \cdot R_k(U) = 0$$

theorem: there exist $n - 1$ independent $v_s^{(k)}(U) \quad \forall k \quad (s = 1 \dots n - 1)$

the $\nabla v_s^{(k)}(U)$ span the orthogonal complement of R_k

5) 'Every constant state is bordered by a simple wave'

k -Riemann Functions in k -simple wave

k -Riemann Functions:

$$\frac{\partial v_s^{(k)}(U)}{\partial U} \cdot R_k(U) \equiv 0$$

• k -simple wave:

$$\frac{\partial U}{\partial t} = \alpha R_k \quad \text{and} \quad \frac{\partial U}{\partial x} = \beta R_k$$

$$\Rightarrow \frac{\partial v_s^{(k)}(U)}{\partial t} = \frac{\partial v_s^{(k)}(U)}{\partial U} \cdot \frac{\partial U}{\partial t} = \alpha \frac{\partial v_s^{(k)}(U)}{\partial U} \cdot R_k = 0$$

$$\Rightarrow \frac{\partial v_s^{(k)}(U)}{\partial t} = \frac{\partial v_s^{(k)}(U)}{\partial U} \cdot \frac{\partial U}{\partial x} = \beta \frac{\partial v_s^{(k)}(U)}{\partial U} \cdot R_k = 0$$

\Rightarrow in k -simple wave the $v_s^{(k)}(U)$ are constant in space and time ($s = 1 \dots n - 1$)

5) *'Every constant state is bordered by a simple wave'*

Property: every constant region is bordered by a characteristic in continuous flow

assume region in (x, t) plane with U constant in space and time, bordered by curve C

- if C is not characteristic \Rightarrow the solution away from C can uniquely be continued
constant continuation is certainly possible \Rightarrow the unique continuation is the constant solution!
 - if the solution is NOT constant away from C, then C has to be a characteristic
(also, C will be a straight characteristic because U is constant on C)
-

5) 'Every constant state is bordered by a simple wave'

Property: every constant region is bordered by a simple wave region

e.g., Lax '57

border curve C is k -characteristic $\Rightarrow k$ -simple wave

$$j\text{-characteristic } x_j(t) : \frac{\partial x_j(t)}{\partial t} = \lambda_j$$

$$L(U) \cdot \frac{\partial U}{\partial t} + \Lambda(U) \cdot L(U) \cdot \frac{\partial U}{\partial x} = 0$$

$$L_j \cdot \left(\frac{\partial U}{\partial t} + \lambda_j \frac{\partial U}{\partial x} \right) = 0 \quad \text{or} \quad L_j \cdot \frac{dU}{dx_j} = 0$$

$$\text{also } L_j \cdot R_k = 0 \quad (j \neq k)$$

$$\text{and thus } L_j = \sum_{s=1}^{n-1} b_{js} \nabla v_s^{(k)}$$

because $\nabla v_s^{(k)}$ span orthogonal complement of R_k

5) 'Every constant state is bordered by a simple wave'

$$\Rightarrow \sum_{s=1}^{n-1} b_{js} \nabla v_s^{(k)} \frac{dU}{dx_j} = 0$$

$$\Rightarrow \sum_{s=1}^{n-1} b_{js} \frac{dv_s^{(k)}}{dx_j} = 0 \quad \forall j \neq k \quad (n-1 \text{ system})$$

$$\Rightarrow \mathbf{B} \cdot \frac{\partial v_s^{(k)}}{\partial t} + \mathbf{\Lambda} \cdot \mathbf{B} \cdot \frac{\partial v_s^{(k)}}{\partial x} = 0$$

$$\Rightarrow \frac{\partial v_s^{(k)}}{\partial t} + \mathbf{B}^{-1} \cdot \mathbf{\Lambda} \cdot \mathbf{B} \cdot \frac{\partial v_s^{(k)}}{\partial x} = 0$$

linear hyperbolic system with $\lambda = \lambda_j \quad (j \neq k)$

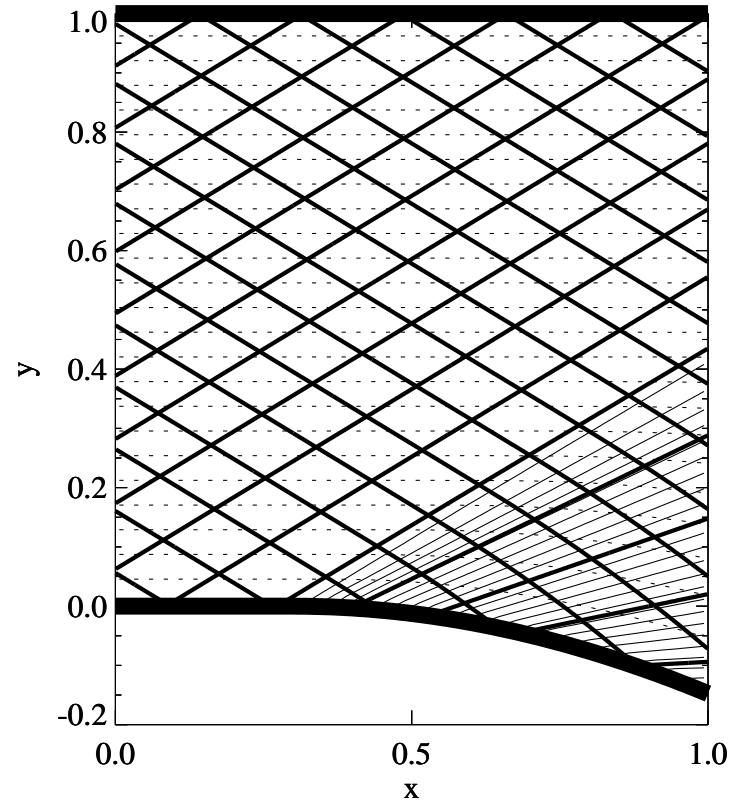
border curve C is not a k -characteristic of this system!

we can uniquely continue the $v_s^{(k)}$ as a constant from the constant region

$v_s^{(k)}$ are constant on the other side of curve C

\Rightarrow the constant region is bordered by a k -simple wave region!

5) *Every constant state is bordered by a simple wave*



5) 'Every constant state is bordered by a simple wave'

this is stated as a general result in textbooks on hyperbolic systems

BUT: is this always true?

NO! what if curve C is a DOUBLE characteristic

$$x_C(t) : \frac{\partial x_C(t)}{\partial t} = \lambda_k = \lambda_{k+1}$$

\Rightarrow proof is not valid

- in Euler: strictly hyperbolic: $v + c \neq v - c$!

this case can never happen; property is generally valid

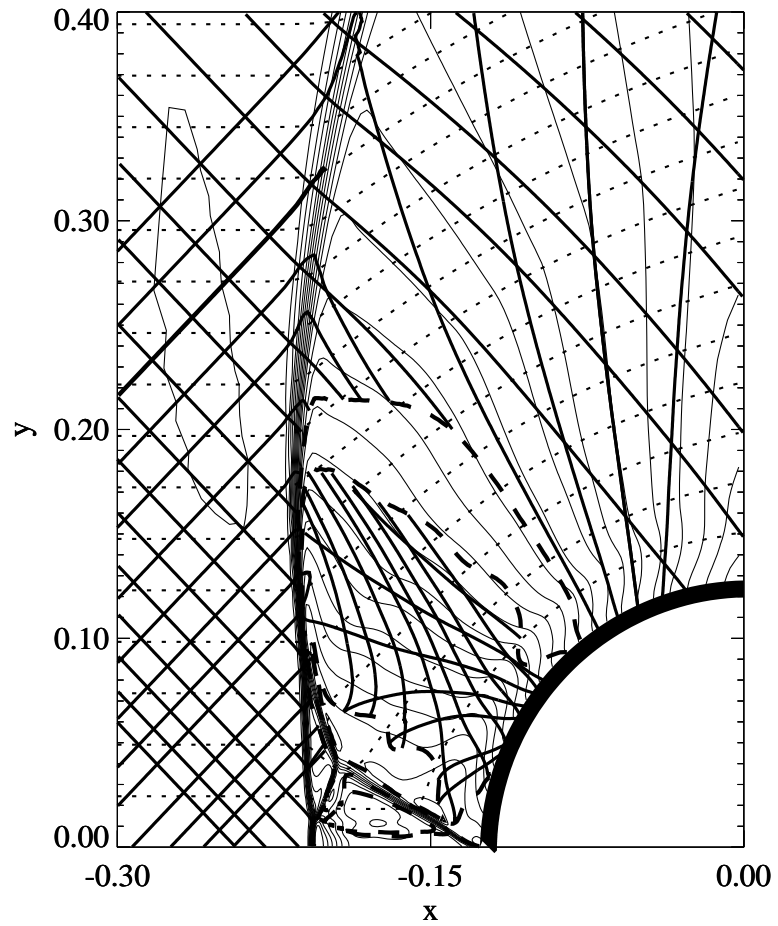
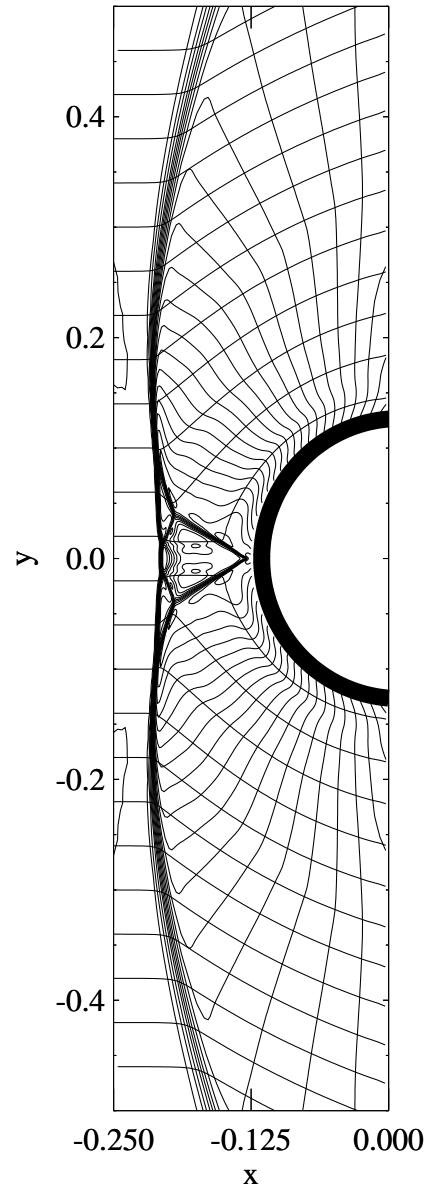
- in MHD: non-strictly hyperbolic: $v + c_s = v + c_f$ is possible

theorem not generally valid, but remains special case

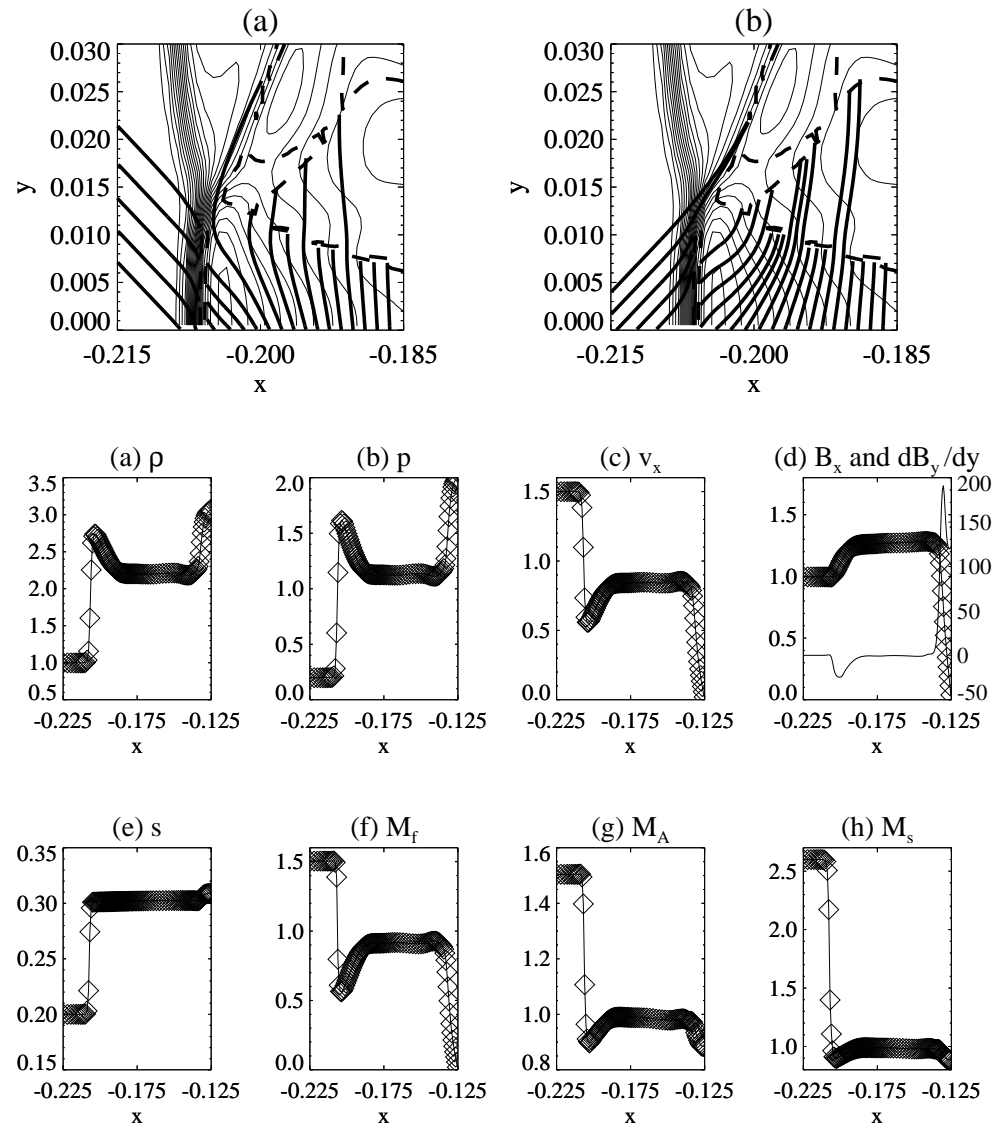
example in 2D simulation results of MHD bow shock

(stationary characteristics \sim space-time characteristics)

5) 'Every constant state is bordered by a simple wave'



5) 'Every constant state is bordered by a simple wave'



Conclusions

- hyperbolic theory of **MHD**:

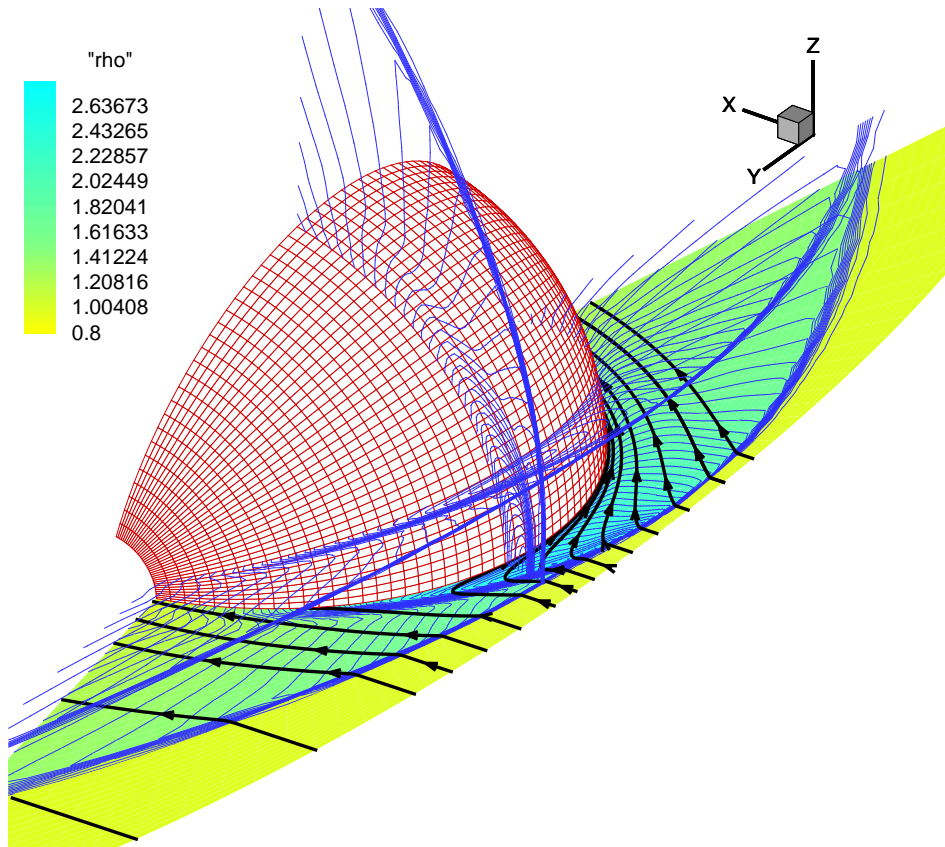
- non-strictly hyperbolic
- non-convex \Rightarrow compound shocks
- rotationally invariant \Rightarrow instability of (overcompressive) intermediate shocks

has been illustrated in 2D and 3D stationary bow shock flows

- compound shocks in 2D and 3D
 - overcompressive (intermediate) shocks in 2D and 3D
 - non-simple wave region bordering constant region
-

Conclusions

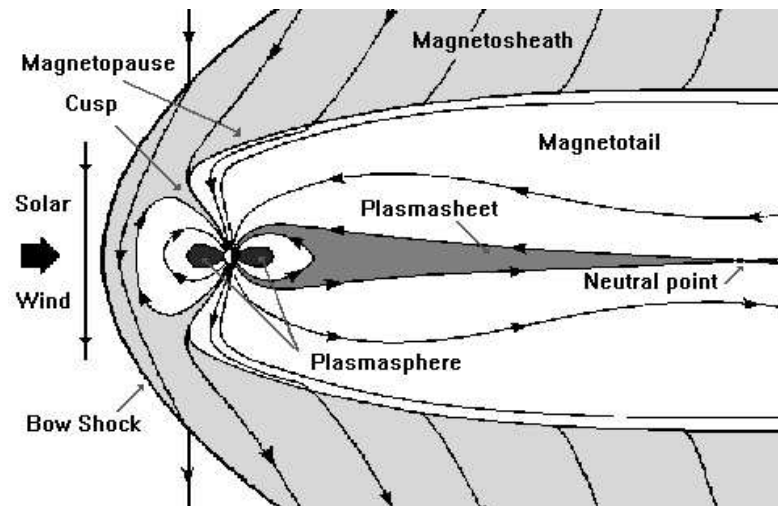
- (overcompressive) intermediate shocks arise naturally in 3D MHD flows
- perturbations destroy shocks, but shocks are reformed in a driven problem



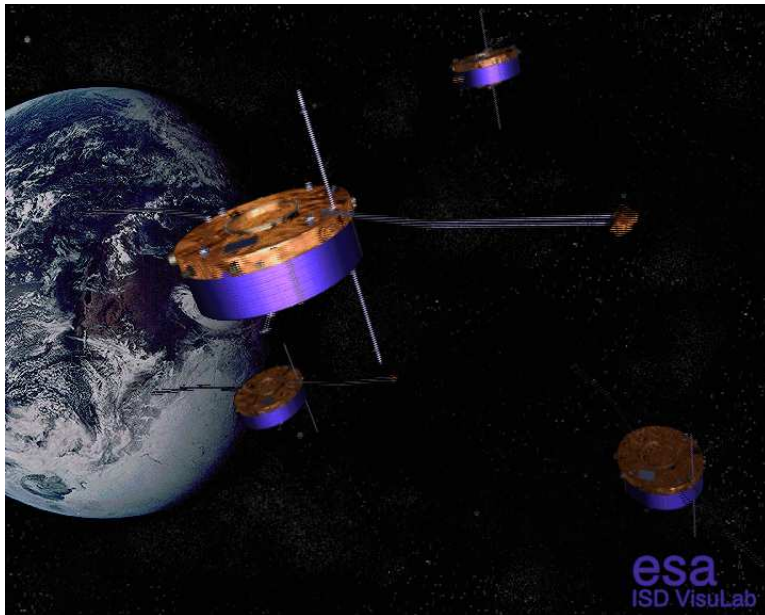
- remaining questions
 - vary dissipation: other solutions?
 - validity of ideal MHD, solution for ideal MHD 3D flow?

Conclusions

- observation of intermediate shocks in Space Physics flows?



CLUSTER II (2000)



STEREO (2003)